



## Brief paper

State-space solution to weight optimization problem in  $\mathcal{H}_\infty$  loop-shaping control<sup>☆</sup>M. Osinuga<sup>a,1</sup>, S. Patra<sup>b</sup>, A. Lanzon<sup>a</sup><sup>a</sup> Control Systems Centre, School of Electrical and Electronic Engineering, University of Manchester, Sackville Street, Manchester, M13 9PL, United Kingdom<sup>b</sup> Department of Electrical Engineering, Indian Institute of Technology, Kharagpur, 721302, India

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## ABSTRACT

This paper proposes a state-space solution to weight optimization problem in  $\mathcal{H}_\infty$  loop-shaping control. A pointwise-in-frequency weight optimization framework is transformed into convex optimization searches that are independent of frequency. The introduced optimization problem therefore avoids gridding of the frequency space and consequently, the inaccuracies attributed to fitting transfer functions to magnitude data. In this optimization problem, the order of the weight is specified 'a priori', thus facilitating the synthesis of low-order controllers, which is desirable from an implementation perspective. The proposed solution algorithm simultaneously synthesizes a robust stabilizing controller and a loop-shaping weight (pre-compensator) that maximize the robust stability margin subject to constraints on the performance and the singular values of the weight. Three numerical examples are given to illustrate the effectiveness of the technique.

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## 1. Introduction

The  $\mathcal{H}_\infty$  loop-shaping design procedure (LSDP), proposed by McFarlane and Glover (1992), is a simple and efficient robust multi-input multi-output (MIMO) controller synthesis technique that combines classical loop-shaping concepts with  $\mathcal{H}_\infty$  synthesis. This well-known design procedure establishes a good tradeoff between robust stability and robust performance of a closed-loop system. Fig. 1 shows this design framework where a stabilizing controller  $C_\infty$  is synthesized for the shaped plant  $P_s = W_2 P W_1$  and the final  $\mathcal{H}_\infty$  loop-shaping controller is obtained by cascading weights with the stabilizing controller  $C_\infty$  as  $C = W_1 C_\infty W_2$ . Typically,  $C$  is of order  $n_p + 2n_w$  (McFarlane & Glover, 1990), where  $n_p$  is the order of  $P$  and  $n_w$  is the combined orders of  $W_1$  and  $W_2$ . This implies that low-order weights facilitate the synthesis of low-order controllers.

In this design procedure, loop-shaping weights  $W_1$  and  $W_2$  are used to shape the singular values of the nominal plant based on the closed-loop performance specifications, and a stabilizing controller that maximizes the robust stability margin is thereafter synthesized. However, weight selection is non-trivial especially

for plants with strong coupling (Hyde, 1995; Lanzon, 2001, 2005; McFarlane & Glover, 1992; Papageorgiou & Glover, 1997; Tsai, Geddes, & Postlethwaite, 1990); factors such as the right-half plane (RHP) poles/zeros of the nominal plant, roll-off rate around crossover, expected bandwidth, singular values and condition numbers of the nominal plant, etc. must also be duly considered. In Lanzon (2001, 2005), these factors have been captured within an optimization framework that maximizes the robust stability margin for a given performance. The solution to the resulting optimization problem simultaneously synthesizes weights and a stabilizing controller.

Although this optimization problem is formulated to hold on a continuum of frequencies, i.e., infinite number of constraints due to the dependence on frequency, it can only be solved by gridding the frequency space. The difficulty in gridding increases with the 'non-smoothness' in the variations of the magnitude response of the plant since the number of grid-points must be chosen such that the intrinsic properties of the plant and the required performance are adequately captured. Furthermore, transfer functions are fitted to the magnitude data obtained from the solution of the pointwise-in-frequency optimization problem to form the loop-shaping weights at each iteration. While high-order transfer functions are required to obtain good curve-fitting for some magnitude data, precise fitting is infeasible in some other cases irrespective of the order of the transfer function, and this might often result in the violation of some of the design constraints. This is due to the fact that curve-fitting depends also on the magnitude response of the plant, in view of the performance requirements. Moreover, fitting, which

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is fixed as the identity and the optimization is over a diagonal pre-compensator  $W_1$ , which is in line with the observation that diagonal loop-shaping weights are generally sufficient to shape the singular values of the nominal plant (Hyde, 1995). The structure and the order of  $W_1$  are specified ‘a priori’ using a parametrization (Sandberg, Lanzon, & Anderson, 2006a,b) which we define next.

#### 4.1. Parametrization of the diagonal pre-compensator

With the aim of formulating a tractable frequency-independent problem, a  $p$ -th order pre-compensator  $W_1 \in \mathcal{GH}_\infty^{n \times n}$  is parametrized as follows:

$$W_1(s) := (a_p s^p + a_{p-1} s^{p-1} + \dots + a_0) \times \left[ \text{diag} \begin{pmatrix} b_p^1 s^p + b_{p-1}^1 s^{p-1} + \dots + b_0^1 \\ \vdots \\ b_p^n s^p + b_{p-1}^n s^{p-1} + \dots + b_0^n \end{pmatrix} \right]^{-1},$$

where coefficients  $a_i, b_i^k \in \mathbb{R} \forall i = 0, 1, \dots, p, k = 1, 2, \dots, n$ .

We define  $\Lambda_{1\omega}$  as follows:

$$\begin{aligned} \Lambda_{1\omega} &= W_1(j\omega)^{-*} W_1(j\omega)^{-1} \\ &= \left( \frac{1}{a_p (-j\omega)^p + a_{p-1} (-j\omega)^{p-1} + \dots + a_0} \right) \\ &\quad \times \left( \frac{1}{a_p (j\omega)^p + a_{p-1} (j\omega)^{p-1} + \dots + a_0} \right) \\ &\quad \times \text{diag} \begin{pmatrix} b_p^1 (-j\omega)^p + b_{p-1}^1 (-j\omega)^{p-1} + \dots + b_0^1 \\ \vdots \\ b_p^n (-j\omega)^p + b_{p-1}^n (-j\omega)^{p-1} + \dots + b_0^n \end{pmatrix} \\ &\quad \times \text{diag} \begin{pmatrix} b_p^1 (j\omega)^p + b_{p-1}^1 (j\omega)^{p-1} + \dots + b_0^1 \\ \vdots \\ b_p^n (j\omega)^p + b_{p-1}^n (j\omega)^{p-1} + \dots + b_0^n \end{pmatrix}. \end{aligned}$$

The above can also be represented as

$$\Lambda_{1\omega} = \left( \frac{1}{\bar{a}_{2p}\omega^{2p} + \bar{a}_{2p-2}\omega^{2p-2} + \dots + \bar{a}_0} \right) \times \text{diag} \begin{pmatrix} \bar{b}_{2p}^{11}\omega^{2p} + \bar{b}_{2p-2}^{11}\omega^{2p-2} + \dots + \bar{b}_0^{11} \\ \vdots \\ \bar{b}_{2p}^{nn}\omega^{2p} + \bar{b}_{2p-2}^{nn}\omega^{2p-2} + \dots + \bar{b}_0^{nn} \end{pmatrix}, \quad (3)$$

where

$$\bar{a}_{2i} := \begin{cases} a_i^2 & \text{if } i = 0, p \\ a_i^2 - \sum_{x=1}^i 2a_{2x-2}a_{2x} \\ \quad + \sum_{x=1}^i 2a_{2x-1}a_{2x+1} & \text{if } i = 1, 2, \dots, p-1, \end{cases}$$

$$\bar{b}_{2i}^{yy} := \begin{cases} (b_i^y)^2 & \text{if } i = 0, p \\ (b_i^y)^2 - \sum_{x=1}^i 2b_{2x-2}^y b_{2x}^y \\ \quad + \sum_{x=1}^i 2b_{2x-1}^y b_{2x+1}^y & \text{if } i = 1, 2, \dots, p-1, \end{cases}$$

$y = 1, 2, \dots, n.$

Furthermore,  $\Lambda_{1\omega}$  in (3) can be expressed in a simplified form as

$$\Lambda_{1\omega} = \mathbf{a}_p^{-1}(\omega^2) B_p(\omega^2),$$

where

$$\mathbf{a}_p(\omega^2) = \bar{a}_{2p}\omega^{2p} + \bar{a}_{2p-2}\omega^{2p-2} + \dots + \bar{a}_0 \quad \text{and}$$

$$B_p(\omega^2) = \text{diag} \begin{pmatrix} \bar{b}_{2p}^{11}\omega^{2p} + \bar{b}_{2p-2}^{11}\omega^{2p-2} + \dots + \bar{b}_0^{11} \\ \vdots \\ \bar{b}_{2p}^{nn}\omega^{2p} + \bar{b}_{2p-2}^{nn}\omega^{2p-2} + \dots + \bar{b}_0^{nn} \end{pmatrix}.$$

Note that  $\mathbf{a}_p(\omega^2)$  and  $B_p(\omega^2)$  are even positive scalar polynomials, where  $\{\bar{a}_k\}$  and  $\{\bar{b}_k^{ii}\}$ ,  $0 \leq k \leq 2p$  and  $1 \leq i \leq n$  are the unknown variables;  $\mathbf{a}_p(\omega^2)$  and  $B_p(\omega^2)$  have  $p$  and  $p \times n$  unknown variables, respectively.

Also, defining the following in view of the state-space formulation:

$$X(s) := \begin{bmatrix} s^p \\ s^{p-1} \\ \vdots \\ 1 \end{bmatrix}, \quad \bar{A}_p := \text{diag} \begin{pmatrix} \bar{a}_{2p} \\ \bar{a}_{2p-2} \\ \vdots \\ \bar{a}_0 \end{pmatrix}, \quad \bar{B}_p := \text{diag} \begin{pmatrix} B_{11} \\ \vdots \\ B_{nn} \end{pmatrix},$$

where

$$B_{ii} = \text{diag} \begin{pmatrix} \bar{b}_{2p}^{ii} \\ \bar{b}_{2p-2}^{ii} \\ \vdots \\ \bar{b}_0^{ii} \end{pmatrix},$$

then, the following can be written:

$$\mathbf{a}_p(\omega^2)I = (I \otimes X(j\omega))^* (I \otimes \bar{A}_p) (I \otimes X(j\omega)), \quad (4)$$

$$\mathbf{a}_p(\omega^2)P^*(j\omega)P(j\omega) = (P(j\omega) \otimes X(j\omega))^* \times (I_n \otimes \bar{A}_p) (P(j\omega) \otimes X(j\omega)), \quad (5)$$

$$B_p(\omega^2) = (I_n \otimes X(j\omega))^* \bar{B}_p (I_n \otimes X(j\omega)). \quad (6)$$

#### 4.2. State-space solution to weight optimization problem

Now, we reformulate the optimization problem stated in Section 3 using the state-space approach and, for simplicity sake, the dependence of  $\mathbf{a}_p$  and  $B_p$  on  $\omega^2$  and  $P, \mathbf{C}, X, \underline{s}, \bar{s}, \underline{w}_1$  and  $\bar{w}_1$  on  $j\omega$  is not shown. We assume that the plant  $P$ , stabilizing controller  $\mathbf{C}$  and the transfer functions  $\underline{s}(s), \bar{s}(s), \underline{w}_1(s)$  and  $\bar{w}_1(s)$  all belong to  $\mathcal{RH}_\infty$  and that  $\underline{s}(s), \bar{s}(s), \underline{w}_1(s), \bar{w}_1(s)$  are single-input single-output (SISO) transfer functions.

##### 4.2.1. Maximizing the robust stability margin over $W_1$

The significance of the robust stability margin has earlier been stated in Section 3 and we now formulate this design objective that maximizes the robust stability margin (by minimizing  $\gamma$ ) as follows:

$$b(P_s, C_\infty) := \left\| \begin{bmatrix} P_s \\ I \end{bmatrix} (I - C_\infty P_s)^{-1} \begin{bmatrix} -C_\infty & I \end{bmatrix} \right\|_\infty^{-1} > \frac{1}{\gamma}.$$

With some algebraic simplifications (Lanzon, 2005) and assuming  $W_2 = I_m$ , the above constraint can equivalently be written as

$$\begin{bmatrix} 0 & P \\ 0 & I \end{bmatrix}^* \begin{bmatrix} I_m & 0 \\ 0 & \Lambda_{1\omega} \end{bmatrix} \begin{bmatrix} 0 & P \\ 0 & I \end{bmatrix} < \gamma^2 \begin{bmatrix} I & P \\ \mathbf{C} & I \end{bmatrix}^* \begin{bmatrix} I_m & 0 \\ 0 & \Lambda_{1\omega} \end{bmatrix} \begin{bmatrix} I & P \\ \mathbf{C} & I \end{bmatrix}. \quad (7)$$

Substituting  $\Lambda_{1\omega}$  with  $\mathbf{a}_p^{-1}B_p$  and multiplying both sides by  $\mathbf{a}_p$  which is a positive scalar polynomial, (7) can be equivalently written as

$$\begin{bmatrix} 0 & P \\ 0 & I \end{bmatrix}^* \begin{bmatrix} \mathbf{a}_p I_m & 0 \\ 0 & B_p \end{bmatrix} \begin{bmatrix} 0 & P \\ 0 & I \end{bmatrix} - \gamma^2 \begin{bmatrix} I & P \\ \mathbf{C} & I \end{bmatrix}^* \begin{bmatrix} \mathbf{a}_p I_m & 0 \\ 0 & B_p \end{bmatrix} \begin{bmatrix} I & P \\ \mathbf{C} & I \end{bmatrix} < 0. \quad (8)$$

Using expressions (4) and (6), constraint (8) can be represented as

$$\begin{bmatrix} 0 & (I_m \otimes X)P \\ 0 & (I_n \otimes X) \end{bmatrix}^* M \begin{bmatrix} 0 & (I_m \otimes X)P \\ 0 & (I_n \otimes X) \end{bmatrix} - \gamma^2 \begin{bmatrix} (I_m \otimes X) & (I_m \otimes X)P \\ (I_n \otimes X) \mathbf{C} & (I_n \otimes X) \end{bmatrix}^* M \begin{bmatrix} (I_m \otimes X) & (I_m \otimes X)P \\ (I_n \otimes X) \mathbf{C} & (I_n \otimes X) \end{bmatrix} < 0, \quad (9)$$

$$\text{where } M = \begin{bmatrix} I_m \otimes \bar{A}_p & 0 \\ 0 & \bar{B}_p \end{bmatrix}.$$

Inequality (9) can also be written as

$$\begin{bmatrix} 0 & (I_m \otimes X)P \\ 0 & (I_n \otimes X) \\ (I_m \otimes X) & (I_m \otimes X)P \\ (I_n \otimes X) \mathbf{C} & (I_n \otimes X) \end{bmatrix}^* \begin{bmatrix} M & 0 \\ 0 & -\gamma^2 M \end{bmatrix} \times \begin{bmatrix} 0 & (I_m \otimes X)P \\ 0 & (I_n \otimes X) \\ (I_m \otimes X) & (I_m \otimes X)P \\ (I_n \otimes X) \mathbf{C} & (I_n \otimes X) \end{bmatrix} < 0. \quad (10)$$

For a given  $P$  and  $\mathbf{C}$ , let  $(A_1, B_1, C_1, D_1)$  be a minimal state-space realization such that

$$\frac{1}{\alpha(s)} \begin{bmatrix} 0 & (I_m \otimes X(s))P(s) \\ 0 & (I_n \otimes X(s)) \\ (I_m \otimes X(s)) & (I_m \otimes X(s))P(s) \\ (I_n \otimes X(s)) \mathbf{C}(s) & (I_n \otimes X(s)) \end{bmatrix} = C_1 (sI - A_1)^{-1} B_1 + D_1, \quad (11)$$

where  $\alpha(s)$  is any  $p$ -th order polynomial with no root on the imaginary axis. Now, dividing (10) by  $\alpha(j\omega)^* \alpha(j\omega)$ , this constraint can then be equivalently expressed as

$$\begin{bmatrix} (j\omega I - A_1)^{-1} B_1 \\ I \end{bmatrix}^* [C_1 \ D_1]^* \begin{bmatrix} M & 0 \\ 0 & -\gamma^2 M \end{bmatrix} \times [C_1 \ D_1] \begin{bmatrix} (j\omega I - A_1)^{-1} B_1 \\ I \end{bmatrix} < 0 \quad \forall \omega. \quad (12)$$

Invoking the KYP lemma (Rantzer, 1996), inequality (12) is equivalent to the existence of a symmetric matrix  $Q_1 \in \mathbb{R}^{n_1 \times n_1}$  such that the following LMI holds

$$\begin{bmatrix} A_1^T Q_1 + Q_1 A_1 & Q_1 B_1 \\ B_1^T Q_1 & 0 \end{bmatrix} + [C_1 \ D_1]^* \begin{bmatrix} M & 0 \\ 0 & -\gamma^2 M \end{bmatrix} [C_1 \ D_1] < 0. \quad (13)$$

Partitioning  $C_1 = \begin{bmatrix} C_{11} \\ C_{12} \end{bmatrix}$  and  $D_1 = \begin{bmatrix} D_{11} \\ D_{12} \end{bmatrix}$ , where the dimensions of the partitioned matrices are compatible with  $M$ , we can write (13) in the form

$$\begin{bmatrix} A_1^T Q_1 + Q_1 A_1 & Q_1 B_1 \\ B_1^T Q_1 & 0 \end{bmatrix} + [C_{11} \ D_{11}]^* M [C_{11} \ D_{11}] < \gamma^2 [C_{12} \ D_{12}]^* M [C_{12} \ D_{12}]. \quad (14)$$

Constraint (14) can be posed as a generalized eigenvalue problem (GEVP) (Boyd, El Ghaoui, Feron, & Balakrishnan, 1994) since

$$M > 0 \quad \text{and} \\ [C_{12} \ D_{12}] \neq 0 \Leftrightarrow [C_{12} \ D_{12}]^* M [C_{12} \ D_{12}] > 0.$$

In view of the objective to maximize the robust stability margin,  $\gamma^2$  is the objective function while  $Q_1$  and  $M$  (i.e.,  $\bar{A}_p$  and  $\bar{B}_p$ ) are the decision variables in this quasiconvex optimization constraint.

#### 4.2.2. Constraints for performance

Next, we formulate constraints (2a) which directly ensure that the desired performance specifications are met by restricting the singular values of the shaped plant within the pre-specified loop-shape boundaries  $\underline{s}$  and  $\bar{s}$  that characterize these specifications:  $|\underline{s}| < \sigma_l(P_s(j\omega)) < |\bar{s}| \forall l$ .

Substituting  $P_s = PW_1$  since  $W_2 = I_m$ , the above can be expressed as  $|\underline{s}|^2 \Lambda_{1\omega} < P^* P < |\bar{s}|^2 \Lambda_{1\omega}$ .

Now, replace  $\Lambda_{1\omega}$  with  $\mathbf{a}_p^{-1}B_p$  in the above expression and multiply by  $\mathbf{a}_p$ :

$$|\underline{s}|^2 B_p < \mathbf{a}_p P^* P < |\bar{s}|^2 B_p.$$

Using the expressions in (5) and (6), and performing several algebraic manipulations similar to the previous formulation, the above constraints, invoking the KYP lemma, are equivalent to the existence of symmetric matrices  $Q_2 \in \mathbb{R}^{n_2 \times n_2}$  and  $Q_3 \in \mathbb{R}^{n_3 \times n_3}$  such that the following LMIs hold

$$\begin{bmatrix} A_2^T Q_2 + Q_2 A_2 & Q_2 B_2 \\ B_2^T Q_2 & 0 \end{bmatrix} + [C_2 \ D_2]^* \begin{bmatrix} \bar{B}_p & 0 \\ 0 & -I_m \otimes \bar{A}_p \end{bmatrix} [C_2 \ D_2] < 0, \quad (15)$$

$$\begin{bmatrix} A_3^T Q_3 + Q_3 A_3 & Q_3 B_3 \\ B_3^T Q_3 & 0 \end{bmatrix} + [C_3 \ D_3]^* \begin{bmatrix} -\bar{B}_p & 0 \\ 0 & I_m \otimes \bar{A}_p \end{bmatrix} [C_3 \ D_3] < 0. \quad (16)$$

Note that  $(A_2, B_2, C_2, D_2)$  and  $(A_3, B_3, C_3, D_3)$  in (15) and (16) are respectively minimal state-space realizations such that

$$\frac{1}{\alpha(s)} \begin{bmatrix} I_n \otimes \underline{s}(s)X(s) \\ P(s) \otimes X(s) \end{bmatrix} = C_2 (sI - A_2)^{-1} B_2 + D_2, \quad (17a)$$

$$\frac{1}{\alpha(s)} \begin{bmatrix} I_n \otimes \bar{s}(s)X(s) \\ P(s) \otimes X(s) \end{bmatrix} = C_3 (sI - A_3)^{-1} B_3 + D_3. \quad (17b)$$

Constraint (15) (resp. (16)) is simultaneously convex in its decision variables  $\bar{A}_p, \bar{B}_p$  and  $Q_2$  (resp.  $Q_3$ ).

#### 4.2.3. Constraints on the singular values of $W_1$

Since a designer needs to ensure that the singular values of the loop-shaping weights are within acceptable bounds, we also consider the constraints on the singular values of  $W_1$  given in (2b) as follows:

$$|\underline{w}_1| < \sigma_l(W_1(j\omega)) < |\bar{w}_1| \quad \forall l \Leftrightarrow |\underline{w}_1|^{-2} I_n > \Lambda_{1\omega} > |\bar{w}_1|^{-2} I_n.$$

Replacing  $\Lambda_{1\omega}$  with  $\mathbf{a}_p^{-1}B_p$  in the above expression, we can then write

$$|\underline{w}_1|^{-2} I_n > \mathbf{a}_p^{-1} B_p > |\bar{w}_1|^{-2} I_n \\ \Leftrightarrow \mathbf{a}_p |\underline{w}_1|^{-2} I_n > B_p > \mathbf{a}_p |\bar{w}_1|^{-2} I_n.$$

Similarly, using (4) and (6) and performing some algebraic manipulations, these inequalities, invoking the KYP lemma, are

equivalent to the existence of symmetric matrices  $Q_4 \in \mathbb{R}^{n_4 \times n_4}$  and  $Q_5 \in \mathbb{R}^{n_5 \times n_5}$  such that the following LMIs hold:

$$\begin{bmatrix} A_4^T Q_4 + Q_4 A_4 & Q_4 B_4 \\ B_4^T Q_4 & 0 \end{bmatrix} + [C_4 \ D_4]^* \begin{bmatrix} \bar{B}_p & 0 \\ 0 & -I_n \otimes \bar{A}_p \end{bmatrix} [C_4 \ D_4] < 0, \quad (18)$$

$$\begin{bmatrix} A_5^T Q_5 + Q_5 A_5 & Q_5 B_5 \\ B_5^T Q_5 & 0 \end{bmatrix} + [C_5 \ D_5]^* \begin{bmatrix} -\bar{B}_p & 0 \\ 0 & I_n \otimes \bar{A}_p \end{bmatrix} [C_5 \ D_5] < 0. \quad (19)$$

Here,  $(A_4, B_4, C_4, D_4)$  and  $(A_5, B_5, C_5, D_5)$  are minimal state-space realizations such that

$$\frac{1}{\alpha(s)} \begin{bmatrix} I_n \otimes \underline{w}_2(s)X(s) \\ I_n \otimes X(s) \end{bmatrix} = C_4 (sI - A_4)^{-1} B_4 + D_4, \quad (20a)$$

$$\frac{1}{\alpha(s)} \begin{bmatrix} I_n \otimes \bar{w}_1(s)X(s) \\ I_n \otimes X(s) \end{bmatrix} = C_5 (sI - A_5)^{-1} B_5 + D_5. \quad (20b)$$

#### 4.2.4. Constraint for the positivity of $\mathbf{a}_p$ and $B_p$

Finally,  $\mathbf{a}_p$  and  $B_p$  are restricted to be positive in order to ensure that  $W_1$  belongs to  $\mathcal{G}_{\mathcal{H}_\infty}$ . These constraints are cast in LMI framework as follows.

$$\mathbf{a}_p = X^* \bar{A}_p X > 0, \quad (21)$$

$$B_p = (I_n \otimes X)^* \bar{B}_p (I_n \otimes X) > 0. \quad (22)$$

If we divide both inequalities (21) and (22) by  $\alpha(j\omega)^* \alpha(j\omega)$  each, and suppose also, minimal state-space realizations  $(A_6, B_6, C_6, D_6)$  and  $(A_7, B_7, C_7, D_7)$  such that

$$\frac{X(s)}{\alpha(s)} = C_6 (sI - A_6)^{-1} B_6 + D_6, \quad (23)$$

$$\frac{I_n \otimes X(s)}{\alpha(s)} = C_7 (sI - A_7)^{-1} B_7 + D_7, \quad (24)$$

constraints (21) and (22) can then be respectively written as

$$\begin{bmatrix} (j\omega I - A_6)^{-1} B_6 \\ I \end{bmatrix}^* [C_6 \ D_6]^* \bar{A}_p \times [C_6 \ D_6] \begin{bmatrix} (j\omega I - A_6)^{-1} B_6 \\ I \end{bmatrix} > 0 \quad \forall \omega, \quad (25)$$

$$\begin{bmatrix} (j\omega I - A_7)^{-1} B_7 \\ I \end{bmatrix}^* [C_7 \ D_7]^* \bar{B}_p \times [C_7 \ D_7] \begin{bmatrix} (j\omega I - A_7)^{-1} B_7 \\ I \end{bmatrix} > 0 \quad \forall \omega. \quad (26)$$

Invoking the KYP lemma, inequalities (25) and (26) are equivalent to the existence of symmetric matrices  $Q_6 \in \mathbb{R}^{n_6 \times n_6}$  and  $Q_7 \in \mathbb{R}^{n_7 \times n_7}$  such that the following LMIs hold

$$\begin{bmatrix} A_6^T Q_6 + Q_6 A_6 & Q_6 B_6 \\ B_6^T Q_6 & 0 \end{bmatrix} - [C_6 \ D_6]^* \bar{A}_p [C_6 \ D_6] < 0,$$

$$\begin{bmatrix} A_7^T Q_7 + Q_7 A_7 & Q_7 B_7 \\ B_7^T Q_7 & 0 \end{bmatrix} - [C_7 \ D_7]^* \bar{B}_p [C_7 \ D_7] < 0.$$

**Remark 1.** An identical  $\alpha(s)$  can be used in the construction of the minimal state-space realizations (11), (17a), (17b), (20a), (20b), (23) and (24) without any loss of generalization. The only requirement is that the  $p$ -th order  $\alpha(s)$  must not have any root on the imaginary axis.

### 4.3. Proposed optimization problem

In order to give the weight optimization problem in state-space characterization, we combine the formulation in the previous subsections as follows:

minimize  $\gamma^2$   
such that  $\exists \mathbf{C} \in \mathcal{C}(P)$ ,  $Q_i \in \mathbb{R}^{n_i \times n_i} \forall i = 1, 2, \dots, 7$

satisfying

$$\begin{bmatrix} A_1^T Q_1 + Q_1 A_1 & Q_1 B_1 \\ B_1^T Q_1 & 0 \end{bmatrix} + [C_{11} \ D_{11}]^* \begin{bmatrix} I \otimes \bar{A}_p & 0 \\ 0 & \bar{B}_p \end{bmatrix} [C_{11} \ D_{11}] < \gamma^2 [C_{12} \ D_{12}]^* \begin{bmatrix} I \otimes \bar{A}_p & 0 \\ 0 & \bar{B}_p \end{bmatrix} [C_{12} \ D_{12}], \quad (27a)$$

$$\begin{bmatrix} A_2^T Q_2 + Q_2 A_2 & Q_2 B_2 \\ B_2^T Q_2 & 0 \end{bmatrix} + [C_2 \ D_2]^* \begin{bmatrix} \bar{B}_p & 0 \\ 0 & -I \otimes \bar{A}_p \end{bmatrix} [C_2 \ D_2] < 0, \quad (27b)$$

$$\begin{bmatrix} A_3^T Q_3 + Q_3 A_3 & Q_3 B_3 \\ B_3^T Q_3 & 0 \end{bmatrix} + [C_3 \ D_3]^* \begin{bmatrix} -\bar{B}_p & 0 \\ 0 & I \otimes \bar{A}_p \end{bmatrix} [C_3 \ D_3] < 0, \quad (27c)$$

$$\begin{bmatrix} A_4^T Q_4 + Q_4 A_4 & Q_4 B_4 \\ B_4^T Q_4 & 0 \end{bmatrix} + [C_4 \ D_4]^* \begin{bmatrix} \bar{B}_p & 0 \\ 0 & -I \otimes \bar{A}_p \end{bmatrix} [C_4 \ D_4] < 0, \quad (27d)$$

$$\begin{bmatrix} A_5^T Q_5 + Q_5 A_5 & Q_5 B_5 \\ B_5^T Q_5 & 0 \end{bmatrix} + [C_5 \ D_5]^* \begin{bmatrix} -\bar{B}_p & 0 \\ 0 & I \otimes \bar{A}_p \end{bmatrix} [C_5 \ D_5] < 0, \quad (27e)$$

$$\begin{bmatrix} A_6^T Q_6 + Q_6 A_6 & Q_6 B_6 \\ B_6^T Q_6 & 0 \end{bmatrix} - [C_6 \ D_6]^* \bar{A}_p [C_6 \ D_6] < 0, \quad (27f)$$

$$\begin{bmatrix} A_7^T Q_7 + Q_7 A_7 & Q_7 B_7 \\ B_7^T Q_7 & 0 \end{bmatrix} - [C_7 \ D_7]^* \bar{B}_p [C_7 \ D_7] < 0, \quad (27g)$$

where  $(A_i, B_i, C_i, D_i)$ ,  $i = 1, 2, \dots, 7$  are minimal state-space realizations of (11), (17a), (17b), (20a), (20b), (23) and (24), respectively. Symmetric matrices  $Q_i \in \mathbb{R}^{n_i \times n_i}$  ( $i = 1, 2, \dots, 7$ ),  $\bar{A}_p$  and  $\bar{B}_p$  are the decision variables in the optimization problem. The optimization problem is quasiconvex when the stabilizing controller  $\mathbf{C} \in \mathcal{C}(P)$  is held fixed in (11) and can be easily solved using LMI routines, for example, 'gevp' solver in MATLAB LMI toolbox.

In this optimization problem that is over a fixed-order bi-proper pre-compensator, inequality (27a) captures the cost function that maximizes the robust stability margin  $b(P_s, C_\infty)$  while (27b)–(27c) delimit the singular values of the shaped plant  $P_s$  within the specified loop-shape boundaries  $\underline{s}(s)$  and  $\bar{s}(s)$  that capture performance specifications. In addition, (27d)–(27e) provide bounds on the singular values of the synthesized pre-compensator while (27f) and (27g) respectively enforce the positivity of  $\bar{A}_p$  and  $\bar{B}_p$  in order to guarantee that the synthesized weight belongs to  $\mathcal{G}_{\mathcal{H}_\infty}$ . We have considered diagonal weight in this formulation, but with slight modification in the parameterizations of the weight as implemented in Sandberg et al. (2006a,b), it is easy to extend the formulation for non-diagonal weight synthesis while still retaining the quasiconvexity of the optimization problem.

**Remark 2.** Constraint on the condition number of  $W_1$  in (2c) has not been directly addressed in the foregoing formulation.

Note that two of the standard closed-loop design objectives are specified as function of the condition number of  $W_1$  (McFarlane & Glover, 1990; Zhou et al., 1996): the gain from plant input disturbance to controller output ( $\bar{\sigma}(C(I + PC)^{-1}P) \leq \min\{\gamma\bar{\sigma}(\tilde{N}_s)\kappa(W_1), 1 + \gamma\bar{\sigma}(M_s)\kappa(W_1)\}$ ) and the gain from plant output disturbance to error signal ( $\bar{\sigma}((I + CP)^{-1}) \leq \min\{1 + \gamma\bar{\sigma}(\tilde{N}_s)\kappa(W_1), \gamma\bar{\sigma}(M_s)\kappa(W_1)\}$ ), both of which should be made small, typically at low frequencies. In the above expressions, the pair  $(\tilde{N}_s, \tilde{M}_s)$  (resp.  $(N_s, M_s) \in \mathcal{RH}_\infty$ ) constitute a normalized left (resp. right) coprime factorization of  $P_s$  (McFarlane & Glover, 1992). At low frequencies,  $\underline{\sigma}(P_s)$  is typically large. Hence,  $\bar{\sigma}(M_s) = \left[\frac{1}{1+\bar{\sigma}(P_s)^2}\right]^{1/2}$  is small, which implies that  $1 + \gamma\bar{\sigma}(M_s)\kappa(W_1)$  and  $\gamma\bar{\sigma}(\tilde{N}_s)\kappa(W_1)$  are made small at these frequencies. Constraint on the condition number of  $W_1$  can therefore be relaxed. However, the choice of  $\underline{w}_1(s)$  and  $\bar{w}_1(s)$  can be made to indirectly ensure that the condition number of  $W_1$  is bounded, especially at low frequencies, i.e., for a given  $\underline{w}_1(s)$  and  $\bar{w}_1(s)$  that satisfies  $\frac{|\bar{w}_1(j\omega)|}{|\underline{w}_1(j\omega)|} < k_1$ , then  $\kappa(W_1(j\omega)) < k_1$ .

$$\left(\bar{\sigma}(W_1) < |\bar{w}_1|, \underline{\sigma}(W_1) > |\underline{w}_1| \Rightarrow \frac{\bar{\sigma}(W_1)}{\underline{\sigma}(W_1)} < k_1 \Leftrightarrow \kappa(W_1) < k_1.\right)$$

## 5. Solution algorithm

A sub-optimal solution algorithm is now proposed based on the state-space formulation given in the previous section. Similar to solution methods for these types of optimization problems (Lanzon & Cantoni, 2003; Lanzon & Tsiotras, 2005; Zhou et al., 1996), for instance,  $D$ - $K$  iterations, the posed problem must be solved iteratively since it is not simultaneously convex in the controller  $C$  and the parameters of the pre-compensator  $W_1$  ( $\bar{A}_p$  and  $\bar{B}_p$ ), used in the construction of the respective minimal state-space realizations.

*Inputs to the algorithm.*

- A scaled nominal plant  $P$ .
- Transfer functions  $\underline{s}(s)$  and  $\bar{s}(s)$  that are boundaries for an allowable loop-shape.
- Transfer functions  $\underline{w}_1(s)$  and  $\bar{w}_1(s)$  that delimit the allowable region for the singular values of  $W_1$ .
- A positive integer  $p$  that specifies the desired order of  $W_1$ .

*The solution algorithm.*

- (1) As a feasible initial starting point for the algorithm, find a controller  $C_0^*$  such that  $[P, C_0^*]$  is internally stable. Set  $i = 0$ , where  $i$  denotes the iteration number and let  $\varepsilon_{\max,0}^* = -1$ .
- (2) Increment  $i$  by 1.
- (3) Formulate and solve the quasiconvex<sup>3</sup> optimization problem in Section 4.3. Denote by  $\bar{A}_p^*$  and  $\bar{B}_p^*$  the values of  $\bar{A}_p$  and  $\bar{B}_p$  that achieve the minimum  $\gamma_i^{2*}$  (denoted by  $\gamma_i^{2*}$ ) of the optimization problem. Furthermore, form  $\Lambda_{1\omega}^* = \mathbf{a}_p(\omega^2)^{-1}B_p(\omega^2)$  using the values of  $\bar{A}_p^*$  and  $\bar{B}_p^*$ .
- (4) Compute a spectral factorization of  $\Lambda_{1\omega}^*$  such that  $W_{1,i}^* \sim W_{1,i}^* = \Lambda_{1\omega}^{*-1}$  and  $W_{1,i}^* \in \mathcal{GH}_\infty$ .
- (5) Compute  $b_{opt}(PW_{1,i}^*)$  as detailed in Glover and McFarlane (1989) and let this value be denoted by  $\varepsilon_{\max,i}^*$ . Furthermore, synthesize a controller  $C_{\infty,i}^*$  that achieves a robust stability margin  $b(PW_{1,i}^*, C_{\infty,i}^*) = \varepsilon_{\max,i}^*$  usually using the state-space formula given in Glover (1984, Theorem 6.3). Set  $C_i^* = W_{1,i}^* C_{\infty,i}^*$ .

- (6) Evaluate  $(\varepsilon_{\max,i}^* - \varepsilon_{\max,i-1}^*)$ . If this difference (which is always positive) is very small, for instance 0.01, and has remained this small for the last few iterations, then EXIT; otherwise return to Step 2.

*Outputs of the algorithm.*

- The maximized value of  $b_{opt}(P_s)$  obtained in the variable  $\varepsilon_{\max,i}^*$ .
- A diagonal loop-shaping weight  $W_{1,i}^*$  that achieves  $\varepsilon_{\max,i}^*$ .
- A controller  $C_{\infty,i}^*(s)$  that achieves  $b(PW_{1,i}^*, C_{\infty,i}^*) = \varepsilon_{\max,i}^*$ .

**Remark 3.** (1) Being an ascent algorithm, the value of  $\varepsilon_{\max,i}^*$  is monotonically non-decreasing as  $i$  increases (due to the existence of weight(s), for instance,  $W_{1,i}^* = I$ , such that  $b(PW_{1,i}^*, W_{1,i-1}^{*-1}C_{i-1}^*) \geq \varepsilon_{\max,i-1}^*$ ) and at each iteration,  $\sqrt{\frac{1}{\gamma_i^{2*}}} \geq \varepsilon_{\max,i-1}^*$ . Note that the algorithm cannot however guarantee convergence to the global minimum; only monotonicity properties can be guaranteed. The use of the algorithm on various practical design examples has however shown that it is quite insensitive to the initial choice of a stabilizing controller  $C_0^*$  and this is most probably due to the fact that it has enough freedom to rectify a poor choice of initial stabilizing controller at both optimization Steps 3 (via  $W_1$ ) and 5 (via  $C_\infty$ ) of each iteration.

(2) Should any of the constraints (27a)–(27g) be required to only hold within a finite frequency range, the finite frequency KYP lemma (Graham, Oliveira, & de Callafon, 2009; Iwasaki, Meinsma, & Fu, 2000) can be invoked in lieu of the KYP lemma Rantzer (1996). Then, the convex constraint will hold accordingly for all frequency  $|\omega| \leq \omega_0$ ,  $\omega_0 \in \mathbb{R}_+$  or  $\omega_1 \leq \omega \leq \omega_2$ , where  $\omega_1$  and  $\omega_2$  are the required lower and upper frequency bounds, respectively.

## 6. Numerical example

The following numerical examples illustrate the effectiveness of the solution algorithm proposed in the previous section. The algorithm has been coded in MATLAB 7.3 on a 2.66 GHz Intel® Core™ 2 PC.

**Example 1.** We consider a lightly-damped (damping ratio of 0.02) SISO plant with poles and zeros at  $s = -0.0283 \pm j1.4139$ ,  $-1.5 \pm j1.6583$  and  $s = -0.4 \pm j19.996$ , respectively. The transfer function of the nominal plant is given below and its magnitude plot is shown in Fig. 2.

$$P = \frac{10(s^2 + 0.8s + 400)}{(s^2 + 0.0566s + 2)(s^2 + 3s + 5)}.$$

Based on the desired closed-loop design objectives, for instance, low frequency gain indicating disturbance rejection and reference signal tracking, bandwidth and roll-off rate around crossover frequency indicating stability and damping, high frequency gain indicating insensitivity to sensor-noise and unmodeled dynamics, etc., the loop-shape boundaries are selected as

$$\underline{s}(s) = \frac{10^3 \left(\frac{s}{0.3} + 1\right) (s^2 + 1.1 \times 10^4 s + 4.65 \times 10^7)}{\left(\frac{s}{0.03} + 1\right)^2 \left(\frac{s}{30} + 1\right)^7};$$

$$\bar{s}(s) = \frac{10^5 \left(\frac{s}{2} + 1\right)^4}{\left(\frac{s}{0.2} + 1\right)^4 \left(\frac{s}{200} + 1\right)^3}.$$

Firstly, we exemplify the problems associated with frequency gridding by using the solution algorithm to the pointwise-in-frequency weight optimization problem stated in Section 3. Frequency functions  $|\underline{w}_1(j\omega)|$  and  $|\bar{w}_1(j\omega)|$  that confine the singular values of  $W_1$  are chosen as  $10^{-2}$  and  $10^2$ , respectively, and the frequency space between  $\omega = 10^{-5}$  and  $10^5$  rad/s is divided

<sup>3</sup> For constraint (27a) in Section 4.3,  $C_{i-1}^*$  is used to obtain the minimal realization of  $(A_1, B_1, C_1, D_1)$  at the  $i$ -th iteration.

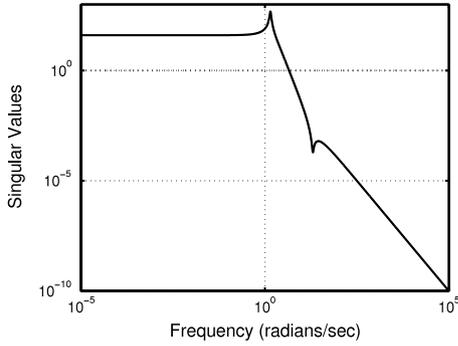


Fig. 2. Magnitude plot of the nominal plant  $P$ .

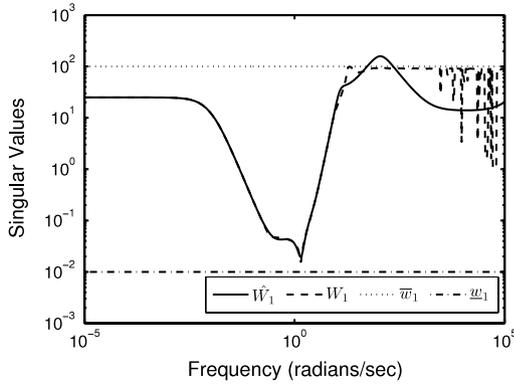


Fig. 3. Curve-fitting using the pointwise-in-frequency weight optimization problem:  $\hat{W}_1$  is the fitted transfer function while  $W_1$  is the obtained magnitude data.

into 500 grid-points. The difficulty in curve-fitting is shown in Fig. 3, where an 11-th order transfer function is fitted to the magnitude data obtained as solution to the optimization problem at the first iteration. There is some considerable inaccuracies in fitting at high frequencies as depicted in the figure. It should be noted that higher-order transfer functions did not improve the accuracy of the curve-fitting. Furthermore, it is observed that the singular values of the synthesized weight do not lie entirely within the specified region due to an inaccurate curve-fitting, i.e., constraints on the singular values of the pre-compensator are violated at certain frequencies, which is undesirable.

Now, the solution algorithm proposed in Section 5 is used to simultaneously synthesize a pre-compensator  $W_1$  and a stabilizing controller  $C_\infty$  that maximize the robust stability margin for the given performance. Since the condition number of SISO system is always 1, the transfer functions  $\underline{w}_1(s)$  and  $\bar{w}_1(s)$ , capturing the bounds on the singular values of the weight, are chosen as gains  $10^{-1}$  and  $10^2$ , respectively. The order  $p$  of the weight is fixed ‘a priori’ as 3. Note that these choices always depend on the particular problem and hence, a different design specification might require designers to select more complicated bounds and/or higher order for the weight. The solution algorithm is now invoked and it practically converges after three iterations. The singular values of the synthesized pre-compensator  $W_1$ , the correspondingly achieved loop-shape  $P_s$  and the singular values of the simultaneously synthesized controller  $C_\infty$  are shown in Fig. 4.

Performance specifications are met in the cascade formed between the nominal plant and the synthesized compensator since the resulting loop-shape lies within the pre-specified region. It is interesting to note that the lightly-damped poles and zeros of the nominal plant are retained in the shaped plant because the synthesized compensator has smooth magnitude response. An otherwise lightly-damped pole-zero cancellation is not desirable as this affects the robust performance and robust stability of

the closed-loop system (see Osinuga, Patra, & Lanzon, 2010). The robust stability margin of 0.4098 is obtained and the order of the synthesized robust stabilizing controller  $C_\infty$  is 6. The synthesized weight  $W_1$  is given in state-space representation below:

$$\begin{bmatrix} -21.7673 & 30.2170 & 0 & 0.2402 \\ -23.6151 & -31.3001 & 0 & 7.9331 \\ -0.0039 & -0.0278 & -0.0028 & -0.0013 \\ -132.9478 & -625.4162 & -24.2412 & 98.6521 \end{bmatrix}$$

**Example 2.** The second example used to demonstrate the applicability of the proposed algorithm is MIMO system taken from process control literature (Loh, Hang, Quek, & Vasnani, 1993), which consists of a 24-tray tower separating methanol and water. The transfer function model for controlling the temperature  $t$  on the 4-th (stripping section) and 17-th (rectifying section) trays is given as

$$\begin{bmatrix} t_{17} \\ t_4 \end{bmatrix} = \begin{bmatrix} -2.2e^{-s} & 1.3e^{-0.3s} \\ 7s + 1 & 7s + 1 \\ -2.8e^{-1.8s} & 4.3e^{-0.35s} \\ 9.5s + 1 & 9.2s + 1 \end{bmatrix} \begin{bmatrix} r \\ s \end{bmatrix},$$

where the manipulated variables are reflux  $r$  and steam-flow  $s$ ; the singular values are shown in Fig. 5. Since the time delays are moderate, they are realized by their 2nd order Padé approximations. Here, the objective is to achieve a unity crossover frequency of approximately 2 rad/s and a good reference signal tracking, i.e., maximized loop-gain at low frequencies. These specifications are approximately captured within the loop-shape boundaries

$$\underline{s}(s) = \frac{2.8 \times 10^7}{(s + 0.02)(s + 80)^4}; \quad \bar{s}(s) = \frac{4.6 \times 10^2}{(s + 0.009)(s + 40)}.$$

Since the pre-compensator cannot roll-off at high frequencies as it is always a bi-proper transfer function, desired roll-off is realized by cascading a low-pass filter with corner frequency of 30 rad/s with the nominal plant. The transfer functions  $\underline{w}_1(s)$  and  $\bar{w}_1(s)$  that confine the singular values of  $W_1$  are chosen as gains of 6.5 and 40, respectively, while the order  $p$  of  $W_1$  is simply chosen as 2 in order to facilitate the synthesis of a low-order controller. The proposed solution algorithm, where performance related constraints (27b) and (27c) are structured to only hold for all frequencies  $|\omega| \leq 10^3$  rad/s by invoking the finite frequency lemma (Iwasaki et al., 2000), is then used to simultaneously synthesize  $W_1$  and  $C_\infty$ .

Accordingly, the singular values of  $W_1$ ,  $P_s$  and  $C_\infty$  at the end of the third iteration when the solution algorithm has practically converged are shown in Fig. 5. It is obvious that the shaped plant lies within the pre-specified loop-shape boundaries, selected based on performance objectives, thus implying that the desired performance specifications are satisfied. The condition number of  $W_1$  is less than 1.3 at all frequencies, which is typically considered good. Finally, and of huge significance, the robust stability margin of 0.3511 is obtained, which is an indicator of a decent design. The order of the synthesized robust stabilizing controller  $C_\infty$  is 14 states and this can be reduced using model reduction techniques given in Zhou et al. (1996), if required. The synthesized weight is given in state-space representation below:

$$\begin{bmatrix} -0.0687 & 0 & 2 & 0 \\ 0 & -0.0760 & 0 & 1 \\ 1.0978 & 0 & 7.1949 & 0 \\ 0 & 1.9573 & 0 & 6.5702 \end{bmatrix}$$

The unit step responses of the system with the corresponding control signals are respectively shown in the last two plots in Fig. 5, where the closed-loop performance is seen to be satisfactory.

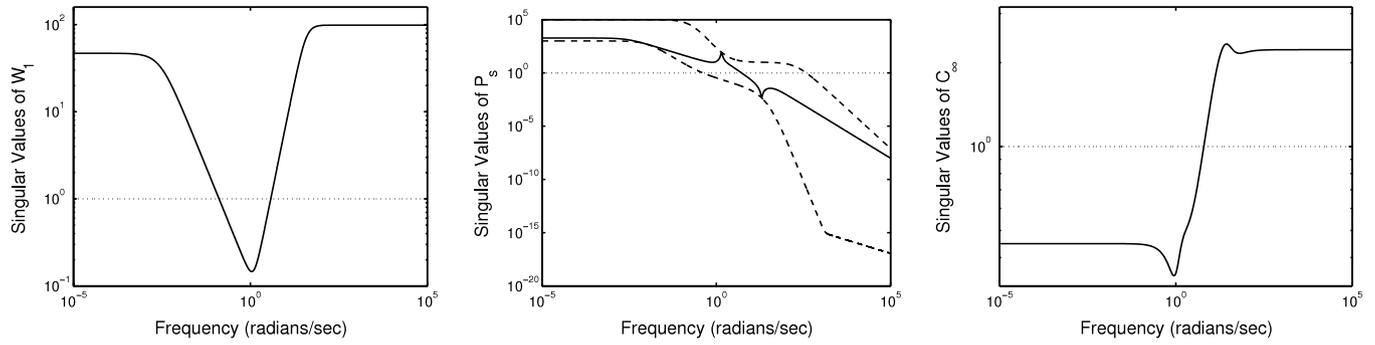


Fig. 4. Singular value plots for  $W_1$ ,  $P_s$  and  $C_\infty$ .

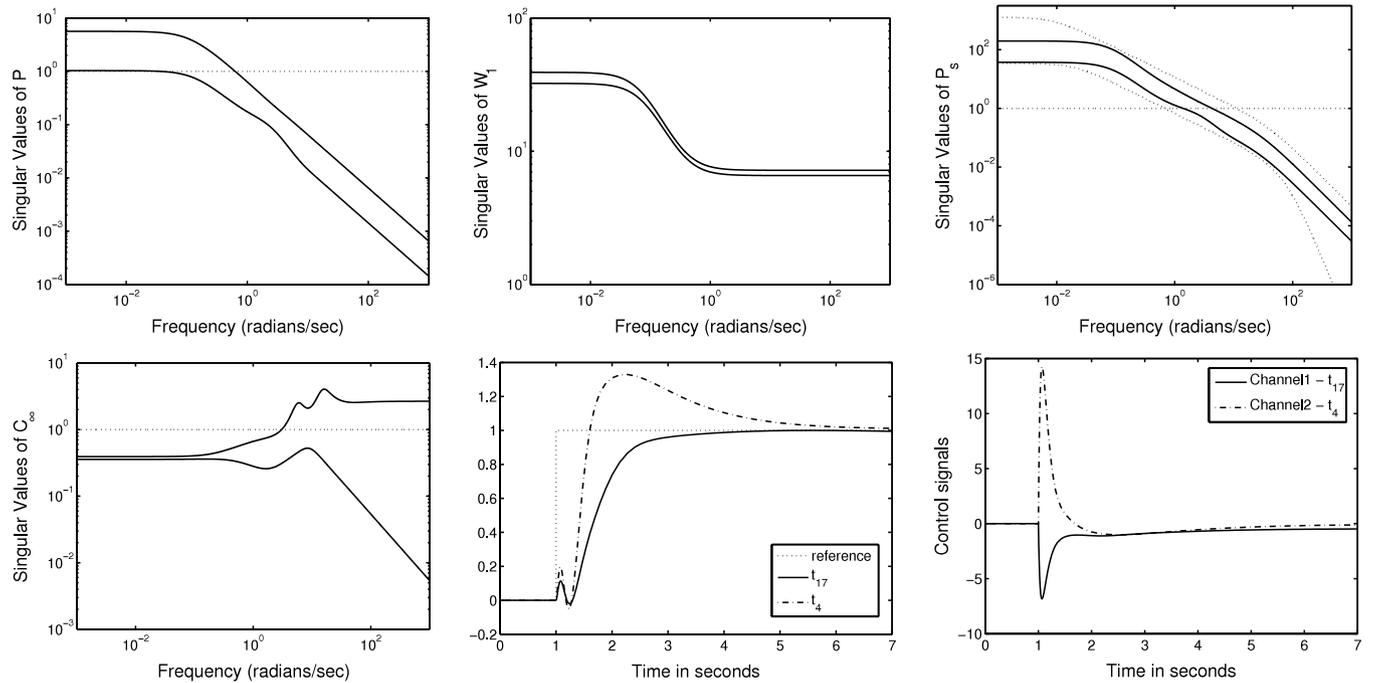


Fig. 5. Singular value plots for  $P$ ,  $W_1$ ,  $P_s$ ,  $C_\infty$  and unit step responses (reflux and steam flow) of the MIMO system and the control signal.

**Example 3.** The same example used in Lanson (2005) is now considered (see the paper for more details). Choosing the order of the pre-compensator as  $p = 2$  and using the same specifications given in the above paper, the resulting weight from the proposed state-space algorithm is given as  $W_1 = \begin{pmatrix} A_{W_1} & B_{W_1} \\ C_{W_1} & D_{W_1} \end{pmatrix}$ , where

$$A_{W_1} = \begin{pmatrix} -0.003 & -1.4594 & 0 & 0 \\ 1.4594 & -748.597 & 0 & 0 \\ 0 & 0 & -0.0021 & -1.5631 \\ 0 & 0 & 1.5631 & -479.7979 \end{pmatrix},$$

$$B_{W_1} = \begin{pmatrix} -10.6547 & 2190.7 & 0 & 0 \\ 0 & 0 & -7.5884 & 1948.7 \end{pmatrix}^T,$$

$$C_{W_1} = \begin{pmatrix} -10.6547 & -2190.7 & 0 & 0 \\ 0 & 0 & -7.5884 & -1948.7 \end{pmatrix}$$

and

$$D_{W_1} = \begin{pmatrix} 6419 & 0 \\ 0 & 7941 \end{pmatrix}.$$

The singular values of this weight and the correspondingly achieved loop-shape are shown in Fig. 6. We obtain a  $b(P_s, C_\infty)$  of 0.3315, which is satisfactory and comparable to the  $b(P_s, C_\infty)$  of 0.368 obtained in Lanson (2005) where the pre-compensator had 8 states.

### 7. Conclusion

This work formulates a weight optimization problem that is dependent on frequency into a finite number of optimization constraints that are independent of frequency. The introduced optimization problem maximizes the robust stability margin over a fixed-order pre-compensator, thus facilitating the synthesis of low-order optimal controllers in  $\mathcal{H}_\infty$  loop-shaping control. The steps of the standard  $\mathcal{H}_\infty$  loop-shaping design procedure are combined in the proposed optimization framework. Consequently, the procedure simultaneously optimizes the robust performance and the robust stability of a closed-loop system in a systematic framework. The algorithm is computationally efficient as the optimization problem is formulated in a quasiconvex framework, which is easily implementable using the available LMI toolbox.

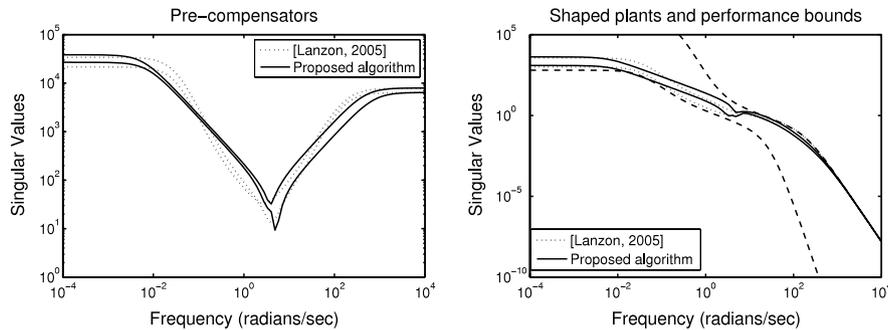


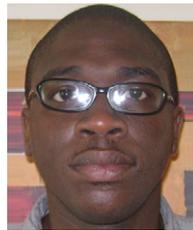
Fig. 6. Loop-shaping weight and shaped plant.

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## References

- Boyd, S., El Ghaoui, L., Feron, E., & Balakrishnan, V. (1994). *Linear matrix inequalities in system and control theory: Vol. 15*. Philadelphia: Society for Industrial and Applied Mathematics.
- Glover, K. (1984). All optimal hankel-norm approximations of linear multivariable systems and their  $\mathcal{L}_\infty$ -error bounds. *International Journal of Control*, 39(6), 1115–1193.
- Glover, K., & McFarlane, D. (1989). Robust stabilization of normalized coprime factor plant descriptions with  $\mathcal{H}_\infty$ -bounded uncertainty. *IEEE Transactions on Automatic Control*, 34(8), 821–830.
- Graham, M., Oliveira, M., & de Callafon, R. (2009). An alternative Kalman–Yakubovich–Popov lemma and some extensions. *Automatica*, 45(6), 1489–1496.
- Hyde, R. (1995).  $\mathcal{H}_\infty$  aerospace control design—a VSTOL flight application. In *Advances in industrial control series*. Berlin: Springer.
- Iwasaki, T., Meinsma, G., & Fu, M. (2000). Generalized S-procedure and finite frequency KYP lemma. *Mathematical Problems in Engineering*, 6(2–3), 305–320.
- Lanzon, A. (2001). Simultaneous synthesis of weights and controllers in  $\mathcal{H}_\infty$  loop-shaping. In *Proceedings of IEEE conference on decision and control, Orlando, FL, US* (pp. 670–675).
- Lanzon, A. (2005). Weight optimisation in  $\mathcal{H}_\infty$  loop-shaping. *Automatica*, 41(1), 1201–1208.
- Lanzon, A., & Cantoni, M. (2003). On the formulation and solution of robust performance problems. *Automatica*, 39(10), 1707–1720.
- Lanzon, A., & Tsiotras, P. (2005). A combined application of  $\mathcal{H}_\infty$  loop-shaping and  $\mu$ -synthesis to control high-speed flywheels. *IEEE Transactions on Control Systems Technology*, 13(5), 766–777.
- Loh, A., Hang, C., Quek, K., & Vasnani, V. (1993). Autotuning of multiloop proportional-integral controllers using relay feedback. *Industrial & Engineering Chemistry Research*, 32(6), 1102–1107.
- McFarlane, D., & Glover, K. (1990). *Lecture notes in control and information sciences: Vol. 138. Robust controller design using normalized coprime factor plant descriptions*. Springer-Verlag.
- McFarlane, D., & Glover, K. (1992). A loop-shaping design procedure using  $\mathcal{H}_\infty$  synthesis. *IEEE Transactions on Automatic Control*, 37(6), 759–769.
- Osinuga, M., Patra, S., & Lanzon, A. (2010). Smooth weight optimization in  $\mathcal{H}_\infty$  loop-shaping design. *Systems and Control Letters*, 59, 663–670.
- Papageorgiou, G., & Glover, K. (1997). A systematic procedure for designing non-diagonal weights to facilitate  $\mathcal{H}_\infty$  loop shaping. In *Proceedings of IEEE conference on decision and control, San Diego, CA, US* (pp. 2127–2132).
- Rantzer, A. (1996). On the Kalman–Yakubovich–Popov lemma. *Systems and Control Letters*, 28(1), 7–10.
- Sandberg, H., Lanzon, A., & Anderson, B. (2006a). Model approximation using magnitude and phase criteria: implications for model reduction and system identification. *International Journal of Robust and Nonlinear Control*, 17(5–6), 435–461.
- Sandberg, H., Lanzon, A., & Anderson, B. (2006b). Transfer function approximation and identification using magnitude and phase criteria. In *Proceedings of the 17th international symposium on mathematical theory of networks and systems, Kyoto, Japan* (pp. 250–256).
- Tsai, M., Geddes, E., & Postlethwaite, I. (1990). Pole-zero cancellations and closed-loop properties of an  $\mathcal{H}_\infty$  mixed sensitivity design problem. In *Proceedings of IEEE conference on decision and control, Honolulu, HI, USA* (pp. 1028–1029).
- Vinnicombe, G. (2001). *Uncertainty and feedback:  $\mathcal{H}_\infty$  loop-shaping and v-gap metric*. Imperial College Press.
- Zhou, K., Doyle, J., & Glover, K. (1996). *Robust and optimal control*. Englewood Cliffs, NJ: Prentice Hall.



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