

## THE WINDSURFER APPROACH TO ITERATIVE CONTROL AND IDENTIFICATION USING AN $\mathcal{H}_\infty$ ALGORITHM

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**Abstract:** The windsurfing approach to iterative control entails several controller designs to gradually and safely widen the closed-loop bandwidth. Eventually some validation tests are carried out in order to stop the algorithm. In this paper, an  $\mathcal{H}_\infty$  design algorithm is presented in order to remove the empirical aspect from the stopping criteria and to make the procedure more systematic, hence expediting the design. Furthermore, a new controller design method is introduced which relaxes some restrictive assumptions on the plant model and tackles well some issues with the controller design step. This enables us to address a wider class of practical problems.

**Keywords:** *Windsurfer Approach; Adaptive Control; Robust Control*

### 1. INTRODUCTION

One of the model-based iterative identification and control design methods, *the windsurfer approach*, was first introduced in (Anderson and Kosut, 1991). This proposes an iterative algorithm which gradually broadens the bandwidth of a closed-loop system through repetition of a two-step procedure using identification and controller re-design steps. Hence, the algorithm, outlined e.g. in (Lee *et al.*, 1993) and (Anderson, 2002), is constructed to achieve the desired performance and design objectives, in which the Internal Model Control (IMC) method (Morari and Zafriou, 1989) is iteratively utilised in the controller design steps. In the case of *stable* plants,

the IMC design method offers the advantage of simplicity and the fact that the closed-loop bandwidth of the system can be set via a single design parameter (Lee *et al.*, 1993). However, if the plant is *unstable*, the same simple IMC design method cannot be used. Treatment of the unstable plant case can be found in papers (Lee *et al.*, 2001) and (Campi *et al.*, 1982). Nevertheless, there are some inadequacies with the IMC design methods for both stable and unstable plants which will be extensively discussed and addressed in section 2.

In this paper, we introduce a modified algorithm for the windsurfer approach in section 3 by incorporating a new controller design method. This new controller design method (section 2.2) maintains the desirable features of the IMC design method, but also extends its applicability and addresses some of the IMC shortcomings. The proposed algorithm introduces an  $\mathcal{H}_\infty$  design technique in order to make the windsurfer approach more systematic and to relax some restrictive assumptions on the plant.

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## 2. CONTROLLER DESIGN METHOD

Internal Model Control is a simple but effective control design method which has been successfully used in many applications where a desired closed-loop amplitude response from reference to plant output must be achieved. Furthermore, for a *stable* plant, adjustment of a single parameter directly and simply determines the closed-loop bandwidth and hence the reference tracking capabilities.

### 2.1 Difficulties with the IMC approach

Consider the feedback system in Fig. 1. One can express  $y$  and  $u$  in terms of  $r_1$  and  $r_2$  as follows:

$$\begin{bmatrix} y \\ u \end{bmatrix} = H(P_i, C_i) \begin{bmatrix} r_1 \\ r_2 \end{bmatrix} = \begin{bmatrix} \frac{P_i C_i}{1+P_i C_i} & \frac{P_i}{1+P_i C_i} \\ \frac{C_i}{1+P_i C_i} & \frac{1}{1+P_i C_i} \end{bmatrix} \begin{bmatrix} r_1 \\ r_2 \end{bmatrix} \quad (1)$$

where  $r_2$  represents a second external input or disturbance acting on the plant input. To motivate the importance of considering the four transfer functions in equation (1), let us note some key points. The (1,1) transfer function  $T_{yr_1} = \frac{P_i C_i}{1+P_i C_i}$  is clearly important for reference tracking and the IMC design method deals particularly with this. Special care, however, must be exercised in regard to the other three transfer functions as the IMC (Morari and Zafiriou, 1989) does not handle them explicitly. For a sensible design, the transfer function from plant input disturbance to output,  $T_{yr_2} = \frac{P_i}{1+P_i C_i}$ , must be maintained below a certain size since we wish plant input disturbances to be attenuated at the plant output. Likewise,  $T_{ur_1} = \frac{C_i}{1+P_i C_i}$  represents the transfer function from reference input to control signal and hence, must be kept below a certain size in order to avoid control actuator saturation and high energy control action. Furthermore, the size of the four transfer functions in equation (1) is related to a generalised robust stability margin (Vinnicombe, 2000) which corresponds to the amount of (coprime factor) uncertainty that can perturb  $P_i$  without destabilising the loop (Zhou *et al.*, 1996). With this introduction in mind, let us now record the circumstances where the IMC design method described in (Morari and Zafiriou, 1989) cannot be properly used or is limited by restrictive assumptions.

- (a) If  $P_i$  is unstable, then the simple IMC design method for the stable case outlined in (Morari and Zafiriou, 1989) cannot be

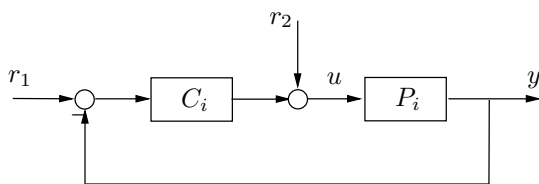


Fig. 1. Standard Feedback Configuration

used. There does exist a different IMC design method for the unstable plant case but the procedure is much more complicated (see (Lee *et al.*, 2001; Campi *et al.*, 1982));

- (b) If  $P_i$  has zeros on the  $j\omega$ -axis, then the factoring of  $P_i$  into stable all-pass,  $[P_i]_a$ , and stable strictly minimum-phase,  $[P_i]_m$ , components as required by (Morari and Zafiriou, 1989) is not possible.
- (c) If the model has lightly-damped stable poles in the closed-loop passband, then  $\frac{P_i}{1+P_i C_i}$  will have large gain near the frequencies of those poles;
- (d) If the model has lightly-damped stable or unstable zeros in the closed-loop passband, then  $\frac{C_i}{1+P_i C_i}$  will have large gain near the frequencies of those zeros;
- (e) If the bandwidth of the IMC filter  $F_i$  (Morari and Zafiriou, 1989) is much larger than the bandwidth of  $P_i$ , then  $|C_i|$  will be very large at frequencies inside the bandwidth of  $F_i$  and outside the bandwidth of  $P_i$ , and again the control signal will be very large;
- (f) If the roll-off rate of  $F_i$  is desired to be less than the roll-off rate of  $P_i$ , then the IMC design method outlined in (Morari and Zafiriou, 1989) will result in an improper controller,  $C_i$ .

Notice that the IMC design method described in (Morari and Zafiriou, 1989) fails and cannot be used in situations (a), (b),(f). However, the IMC design method can be applied in situations (c),(d) and (e) but difficulties may occur. In the sequel, we shall introduce a new controller design method that inherits the useful desired features of the IMC design method, but addresses problems stated in situations (a)–(f) explicitly.

### 2.2 Proposed $\mathcal{H}_\infty$ Controller Design Method

On the one hand, we would like to utilise the desired features of the IMC on either a stable or unstable plant. On the other hand, we would like to ensure that the magnitudes of all the transfer function matrix entries in equation (1) do not become too large. These two objectives, however, are not the same and we need to reformulate the problem in such a way to capture our objectives. Therefore, we shall introduce an  $\mathcal{H}_\infty$  index and require this index to be minimised over all stabilising controllers. We normally have performance objectives in mind, which requires some transfer functions to be small or below certain values in some frequency regions and other transfer functions small or below certain values at other frequencies. The  $\mathcal{H}_\infty$  index will be weighted to achieve the desired effect. Let us outline our proposed controller design method:

- (i) Given a model of the plant,  $P_i$  do the following factorisation:

$$P_i = [P_i]_a [P_i]_m ; \begin{cases} [P_i]_a \in \mathcal{RH}_\infty, [P_i]_a^\sim [P_i]_a = I \\ [P_i]_m \text{ has no zeros in } \mathbb{C}_+ \end{cases} \quad (2)$$

where  $\mathbb{C}_+$  denotes the open right-half plane.

- (ii) The admissible controller is given by solving the following  $\mathcal{H}_\infty$  problem:

$$\gamma_i = \min_{C_i \in \underline{\mathcal{C}}} \left\| \begin{bmatrix} \frac{P_i C_i}{1+P_i C_i} - [P_i]_a F_i & \epsilon_2(s) \frac{P_i}{1+P_i C_i} \\ \epsilon_1(s) \frac{C_i}{1+P_i C_i} & \epsilon_1(s) \epsilon_2(s) \frac{1}{1+P_i C_i} \end{bmatrix} \right\|_\infty \quad (3)$$

where  $\underline{\mathcal{C}}$  denotes the set of all proper stabilising controllers for the plant,  $P_i$ , and  $\epsilon_1(s)$  and  $\epsilon_2(s)$  are SISO, stable, minimum-phase and proper weights<sup>3</sup>. We shall explain in detail the selection of the weighting functions in the following section.

One can easily verify that this proposed design method addresses all the aforesaid difficulties, cases (a)–(f), with the IMC design method.

### 2.3 Design of weighting functions $\epsilon_1(s)$ and $\epsilon_2(s)$

The weighting functions were introduced as a part of the  $\mathcal{H}_\infty$  index in equation (3) and we shall now discuss the way to design them. One should realize that, based upon the particular application specifications and also the characteristics of the plant, we will have different objectives in different frequency regions. Let us now set out our design objectives, as specified in index (3):

- (1) Let  $\alpha$  be the desired closeness between  $\frac{P_i C_i}{1+P_i C_i}$  and  $[P_i]_a F_i$  in an  $\mathcal{H}_\infty$  sense. That is, we require  $\left\| \frac{P_i C_i}{1+P_i C_i} - [P_i]_a F_i \right\|_\infty \leq \alpha$ .
- (2) Let  $\beta_p$  be the maximum tolerable gain in the appropriate frequency region for the transfer function  $T_{yr2} = \frac{P_i}{1+P_i C_i}$ . That is, we require  $\bar{\sigma} \left[ \frac{P_i}{1+P_i C_i}(j\omega) \right] \leq \beta_p \quad \forall \omega \in [\omega_1, \omega_2]$ .
- (3) Let  $\beta_c$  be the maximum tolerable gain in the appropriate frequency region for the transfer function  $T_{ur1} = \frac{C_i}{1+P_i C_i}$ . That is, we require  $\bar{\sigma} \left[ \frac{C_i}{1+P_i C_i}(j\omega) \right] \leq \beta_c \quad \forall \omega \in [\omega_3, \omega_4]$ .

Now, we have three different numbers, i.e.  $\alpha$ ,  $\beta_p$  and  $\beta_c$ , that capture our objectives. These three numbers will be used to specify  $\epsilon_1(s)$  and  $\epsilon_2(s)$  as we discuss next. Once  $\epsilon_1(s)$  and  $\epsilon_2(s)$  are specified, we just need to check the number  $\gamma_i$  to determine whether the design was successful in achieving our objectives or not. Towards this end, note that the index in (3) certainly guarantees that:

$$\bar{\sigma} \left[ \frac{P_i C_i}{1+P_i C_i} - [P_i]_a F_i \right] \leq \gamma_i \quad \forall \omega \quad , \quad (4)$$

$$\bar{\sigma} \left[ \frac{P_i}{1+P_i C_i}(j\omega) \right] \leq \frac{\gamma_i}{|\epsilon_2(j\omega)|} \quad \forall \omega \quad , \quad (5)$$

$$\bar{\sigma} \left[ \frac{C_i}{1+P_i C_i}(j\omega) \right] \leq \frac{\gamma_i}{|\epsilon_1(j\omega)|} \quad \forall \omega \quad (6)$$

are achieved. Let us consider four different scenarios that describe how  $\epsilon_1(j\omega)$  and  $\epsilon_2(j\omega)$  ought to be chosen.

<sup>3</sup> Note that  $\min_{C_i \in \underline{\mathcal{C}}} \|\mathcal{F}_l(\cdot, \cdot)\|_\infty$  rather than  $\inf_{C_i \in \underline{\mathcal{C}}} \|\mathcal{F}_l(\cdot, \cdot)\|_\infty$  is used since we need both  $\gamma_i$  and  $C_i \in \underline{\mathcal{C}}$ .

**2.3.1. If  $\epsilon_1(j\omega) = 0$  and  $\epsilon_2(j\omega) = 0$ ,** then the  $\mathcal{H}_\infty$  index specified in equation (3) reduces to

$$\gamma_i = \min_{C_i \in \underline{\mathcal{C}}} \left\| \frac{P_i C_i}{1+P_i C_i} - [P_i]_a F_i \right\|_\infty \quad (7)$$

and hence  $C_i$  will be exactly the standard IMC controller<sup>4</sup>, at least if  $P_i$  is stable and has no  $j\omega$ -axis zeros and all other assumptions of the IMC design method outlined in (Morari and Zafiriou, 1989) are fulfilled (i.e.  $\gamma_i = 0$  for such a case)<sup>5</sup>. Thus,  $\epsilon_1(j\omega)$  and  $\epsilon_2(j\omega)$  can be set to be very small in the frequency regions where the plant characteristics and our performance objectives are such that an IMC controller can perform well at those frequencies.

**2.3.2. If  $\epsilon_1(j\omega) = 0$  but  $\epsilon_2(j\omega) \neq 0$ ,** then it is clear that, in this situation, we are trying to make  $\frac{P_i C_i}{1+P_i C_i}$  close to  $[P_i]_a F_i$  but simultaneously we are seeking to limit the size of  $\frac{P_i}{1+P_i C_i}$ . With reference to section 2.1, we choose  $|\epsilon_2(j\omega)| \geq \frac{\alpha}{\beta_p}$  near the frequencies of the lightly-damped poles of  $P_i$ , as this will then limit the size of  $\frac{P_i}{1+P_i C_i}$ .

**2.3.3. If  $\epsilon_1(j\omega) \neq 0$  but  $\epsilon_2(j\omega) = 0$ ,** then it is clear that, in this situation, we are trying to make  $\frac{P_i C_i}{1+P_i C_i}$  close to  $[P_i]_a F_i$  but limit the size of  $\frac{C_i}{1+P_i C_i}$ . With reference to section 2.1, we choose  $|\epsilon_1(j\omega)| \geq \frac{\alpha}{\beta_c}$  near the frequencies of the lightly-damped zeros of  $P_i$ , as this will limit the size of  $\frac{C_i}{1+P_i C_i}$ .

**2.3.4. If  $\epsilon_1(j\omega) \neq 0$  and  $\epsilon_2(j\omega) \neq 0$ ,** then we can trade-off the closeness requirement of  $\frac{P_i C_i}{1+P_i C_i}$  to  $[P_i]_a F_i$  with limiting the size of both  $\frac{P_i}{1+P_i C_i}$  and  $\frac{C_i}{1+P_i C_i}$  at the appropriate frequencies. Again,  $\epsilon_1(j\omega)$  and  $\epsilon_2(j\omega)$  are chosen such that  $|\epsilon_1(j\omega)| \geq \frac{\alpha}{\beta_c}$  and  $|\epsilon_2(j\omega)| \geq \frac{\alpha}{\beta_p}$  at the right frequencies.

## 3. NEW WINDSURFER DESIGN METHOD

The iterative control and identification approach to adaptive control includes two types of steps: identification and controller design steps. With a stabilising controller acting on the true plant, a model of the plant is identified through a closed-loop identification procedure.

As mentioned in the introduction, the originally proposed windsurfer approach to iterative identification and control re-design (Lee *et al.*, 1993)

<sup>4</sup> Note though that the  $\mathcal{H}_\infty$  index can be used even if the plant,  $P_i$ , is unstable or has  $j\omega$ -axis zeros, although  $\gamma_i$  may not be equal to zero in this case.

<sup>5</sup> The minimum may not be attainable when  $P_i$  is unstable or has  $j\omega$ -axis zeros or the roll-off rate of  $F$  is smaller than that of  $P_i$  in which case there would be an infimum but not a minimum.

gradually and safely increases the closed-loop bandwidth using the IMC design method at the control stage. As extensively discussed in section 2.1, the IMC design method suffers from some serious deficiencies. These are, however, all addressed in our new  $\mathcal{H}_\infty$  design method (see section 2.2). Hence, in the sequel, we shall propose a *modified* windsurfer algorithm that makes use of our new  $\mathcal{H}_\infty$  design method, instead of the IMC, at the control design step. Because of this change, we modify other parts of the windsurfer algorithm to make it more amenable to our design method. It is important to understand that our new  $\mathcal{H}_\infty$  design method is better suited to flag when the re-identification is required, when compared to the IMC design method, as will become evident below.

### 3.1 A Modified Iterative Identification and Controller Re-design Algorithm

Given a true physical plant,  $P_t$ , which is unknown, and a known controller,  $C_0$ , which stabilises  $P_t$ , the algorithm follows the steps outlined below.

- **Step 1.** Perform the closed-loop identification, outlined in (Anderson, 2002) and the references therein, so that the identified plant model  $P_i$  satisfies:  $[P_i, C_i]$  stable and

$$\max_{1 \leq k \leq N} \bar{\sigma}[H(P_t, C_i) - H(P_i, C_i)(j\omega_k)]$$

small<sup>6</sup>.

- **Step 2.** Set an initial bandwidth,  $\lambda_0$ , for the desired closed-loop response,  $F_i(j\omega)$ . The initial bandwidth can be chosen to be much smaller than that of the identified model where its magnitude response is flat. It is always safe to choose the initial bandwidth of  $F_i(j\omega)$  to be an order of magnitude less than the lowest predominant feature of  $P_i(j\omega)$ . That is, if there is a resonant pole at  $\omega_0$  inside the plant bandwidth, choose the initial bandwidth for  $F_i(j\omega)$  to be  $\frac{\omega_0}{10}$ ;
- **Step 3.** Find the critical frequency regions in  $P_i$  based on the cases, (a) – (f), discussed in section 2.1. Based on the desired closed-loop objectives and specifications, set:
  - (a) the positive number  $\alpha$  in the sense that  $\left\| \frac{P_i C_i}{1 + P_i C_i} - [P_i]_a F_i \right\|_\infty \leq \alpha$ .
  - (b) the positive numbers  $\beta_p$  and  $\beta_c$  in the sense that  $\begin{cases} \bar{\sigma} \left[ \frac{P_i}{1 + P_i C_i}(j\omega) \right] \leq \beta_p & \forall \omega \in [\omega_1, \omega_2] \\ \bar{\sigma} \left[ \frac{C_i}{1 + P_i C_i}(j\omega) \right] \leq \beta_c & \forall \omega \in [\omega_3, \omega_4] \end{cases}$
- **Step 4.** Design the frequency weights,  $\epsilon_1(s)$  and  $\epsilon_2(s)$ , according to the rules given in section 2.3, using the specified values  $\alpha$ ,  $\beta_p$  and  $\beta_c$  in **Step 3**, for the appropriate frequency regions;

- **Step 5.** Solve the  $\mathcal{H}_\infty$  controller design problem given in equation (3) and obtain  $\gamma_i$  and  $C_i$ ;
- **Step 6.** If  $\gamma_i \leq \alpha$  and  $\max_{1 \leq k \leq N} \bar{\sigma}[H(P_t, C_i) - H(P_i, C_i)(j\omega_k)]$  is small, then increase the bandwidth for  $F_i$  by 10%; and go to **Step 5**. Otherwise, go to **Step 7**;
- **Step 7.** Discard the last controller. IF there was a previous controller, then go to **Step 1**, else **END**.

## 4. NUMERICAL EXAMPLE

In this section, we consider a plant which is examined in (Lee *et al.*, 1993) and (Anderson, 2002) in order to discuss the proposed algorithm and also to show its systematic easy-to-use features.

The true plant is a robot arm with the transfer function<sup>7</sup>,  $P_t(s) = 0.5196 \frac{\prod_{i=1}^5 (s - z_i)}{\prod_{i=1}^6 (s - p_i)}$  where the zeros are  $-13.162$ ,  $-10.646 \pm j12.27$ ,  $7.169 \pm j11.54$  and the poles are  $-0.0996 \pm j3.0017$ ,  $-0.3339 \pm j12.131$ ,  $-1.8450 \pm j31.481$ .

We shall apply the algorithm described in section 3.1 and use the closed-loop identification procedure detailed in (Hansen, 1989) and outlined in (Anderson, 2002). To start, the identification procedure requires an initial model of the plant and a controller that stabilises both the model and the true plant. The initial model  $P_0$ , obtained from (Lee *et al.*, 1995), and the controller  $C_0$ , standard IMC controller to achieve a closed-loop bandwidth of 0.1 rad/s, are chosen to be

$$P_0 = \frac{0.1709(s+13.31)}{s^2+0.1806s+9.024}, \quad C_0 = \frac{0.5849(s^2+0.1806s+9.024)}{s(s+13.31)}.$$

Using the identification procedure of (Hansen, 1989), we identify a model  $\tilde{P}_1$  using  $P_0$  and  $C_0$  and the identification data collected. The (reference) input is chosen to consist of four periods of a zero-mean square wave of amplitude 1. The plant output was corrupted by zero-mean white noise with standard deviation of 0.05. The Numerical Subspace-Based State Space Model Estimation N4SID (built in MATLAB function) is used to estimate the model. Choosing the identified Youla-Kucera parameter  $R$  (discussed in (Anderson, 2002)) to be of order five, the resulting identified model,  $\tilde{P}_1$ , is of degree eleven and is given by  $\tilde{P}_1(s) = 0.51819 \frac{1.0 \times 10^2 \prod_{i=1}^7 (s - z_i)}{1.0 \times 10^2 \prod_{i=1}^8 (s - p_i)}$  where the zeros are  $-0.0038$ ,  $-0.1331$ ,  $-8.9357$ ,  $-0.0043$ ,  $0.0628 \pm j0.1300$ ,  $-0.0658 \pm j0.0802$ ,  $-0.1573 \pm j0.0109$  and the poles are  $-0.0039$ ,  $-0.0041$ ,  $-8.9357$ ,  $-0.0297 \pm j0.3068$ ,  $-0.0070 \pm j0.1180$ ,  $-0.1130 \pm j0.0030$ ,  $-0.0009 \pm j0.0300$ .

For this identified plant model  $\tilde{P}_1$ ,  $\max_{1 \leq k \leq N} \bar{\sigma}[H(\tilde{P}_1, C_0) - H(P_t, C_0)(j\omega_k)] = 0.4683$ . Although one may think that 0.4683 is not small enough, we point out

<sup>6</sup> It can be shown that if  $[P_t, C_i]$  and  $[P_i, C_i]$  are stable then for SISO systems  $\bar{\sigma}[H(P_t, C_i) - H(P_i, C_i)] = \left[ |C_i| + \frac{1}{|C_i|} \right] \left| \frac{P_t C_i}{1 + P_t C_i} - \frac{P_i C_i}{1 + P_i C_i} \right|$ .

<sup>7</sup> Note that, the windsurfer approach does not have available to it the mathematical description of the true plant; we only use this information to verify the method.

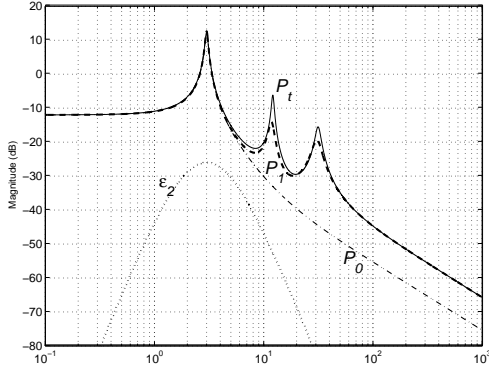


Fig. 2. Magnitude Responses of  $P_t$ ,  $P_0$ ,  $P_1$ , and  $\epsilon_2$

that (a) for the identified model  $G_1$  in (Lee *et al.*, 1995)  $\max_{1 \leq k \leq N} \bar{\sigma}[H(P_t, C_i) - H(G_1, C_i)(j\omega_k)] = 0.8637$  so our scheme is no worse; (b) it is very difficult for identification schemes to pick out the detailed resonant structures of an unknown true plant, and typically the largest identification errors occur at or near these resonances.

To avoid degree explosion in the identified model, we employ the closed-loop model reduction method detailed in section 4.3 (pp 137–140) of reference (Obinata and Anderson, 2001). The Hankel singular values of the graph symbols of  $\hat{P}_1$  are 2.1976, 2.0657, 1.5216, 0.11324, 0.1191, 0.0801, 0.0552, 0.0460, 0.0002,  $2.84e-5$ ,  $7.38e-10$ . We perform balanced realization and the result is truncated to retain all Hankel singular values greater than  $0.01\sigma_1$ . The identified plant  $P_1$  is stabilised by  $C_0$ , and we estimate  $\max_{1 \leq k \leq N} \bar{\sigma}[H(P_t, C_0) - H(P_1, C_0)(j\omega_k)] = 0.4684$ . The magnitude responses of the true plant  $P_t(j\omega)$  and the model  $P_1(j\omega)$  are shown in Fig. 2.

The initial bandwidth,  $\lambda_1$  is set to 0.1, which is much smaller than that of  $P_1$  and is well below the first resonant frequency. We divide (section 2.3) the plant model frequency response into two regions; *region 1*: below 1.5 *rad/s*, and *region 2*: contains the frequencies around the first resonant mode. We shall now set our closed-loop performance objectives to be: (i)  $\alpha = 0.1$ ; (ii)  $\beta_p = 2$  ( $\approx 6$  dB). Here, the closed-loop bandwidth  $\lambda_1$  lies in the *first region*, hence  $\epsilon_1(j\omega)$  is set to be small (say 0.001), and  $\epsilon_2(j\omega)$  is chosen to take care of the first resonant mode of  $P_1$  at 3 *rad/s*;  $\epsilon_2(j\omega)$  is set, Fig. 2, to have the maximum gain of  $-26$  dB, ( $\alpha/\beta_p = 0.1/2 = 0.05$ ).

As stated in **Step 5** of our algorithm, we then solve the  $\mathcal{H}_\infty$  index in equation (3) with  $F_1 = \frac{\lambda_1}{s + \lambda_1}$  and the controller obtained achieves the norm value of 0.10 ( $\gamma_1 = 0.10 \leq \alpha$ ). The second condition in **Step 6** of our algorithm,  $\max_{1 \leq k \leq N} \bar{\sigma}[H(P_t, C_0) - H(P_1, C_0)(j\omega_k)] = 0.3136$ . The two conditions of **Step 6** are met until we push out the bandwidth to  $\lambda_i = 1.0$  where  $\gamma_i = 0.09 < \alpha$  but the second condition is not satisfied since the norm

has increased to a value (0.4698) greater than the initial one (0.4684). This points out that re-identification is necessary. Therefore, we discard the last controller and go back to **Step 1** for re-identification.

After re-identification, we can gradually (10% in each iteration) increase the closed-loop bandwidth  $\lambda_i$  to 8.2 *rad/s* where both conditions of **Step 6** are not satisfied and hence re-identification is required.

We perform the re-identification again and then set our closed-loop performance objectives to be: (i)  $\alpha = 0.1$ ; (ii)  $\beta_p = 2$  ( $\approx 6$  dB); (iii)  $\beta_c = 30$  ( $\approx 30$  dB). The frequency cost  $\epsilon_1(j\omega)$  is chosen to gradually reach the maximum gain of almost  $-50$  dB ( $\alpha/\beta_c = 0.1/30$ ) at 4 *rad/s*, where the plant model loses its bandwidth to the controller. The cost  $\epsilon_2(j\omega)$  is chosen to be the same as before (Fig. 3). Now, The closed-loop bandwidth can be further pushed out to 11.2 *rad/s*. After re-identification, the bandwidth can not be increased more than 11.2 *rad/s* as  $\gamma_i$  will be more than 0.1 and the performance objectives will not be met.

Let us now record a number of key attributes which make our proposed algorithm much more suitable for the windsurfer approach. The results in (Lee *et al.*, 1995) show that the closed-loop bandwidth can be expanded to 12 *rad/s* whereas ours can be expanded to 11.2 *rad/s*. Considering the 4-block transfer function  $H(G_1, C_{IMC})$ , where  $G_1$  denotes the identified model in (Lee *et al.*, 1995) and  $C_{IMC}$  refers to the controller which was designed to achieve the close-loop bandwidth of 12 *rad/s*, we note the following important points:

- the magnitude response of the transfer function  $\frac{G_1 C_{IMC}}{1 + G_1 C_{IMC}}$  is exactly specified by the IMC filter  $F = \frac{12}{(s+12)^2}$  and hence exact reference tracking is achieved for the desired bandwidth on the identified model. However,  $\max_{1 \leq k \leq N} \bar{\sigma}[\frac{P_t C_{IMC}}{1 + P_t C_{IMC}} - \frac{G_1 C_{IMC}}{1 + G_1 C_{IMC}}(j\omega_k)] = 0.57$  which indicates a large discrepancy in reference

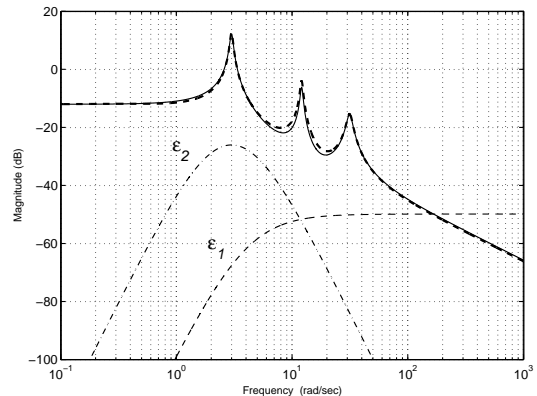


Fig. 3. Magnitude Responses of  $P_t$ ,  $P_3$ ,  $\epsilon_1$  and  $\epsilon_2$

tracking capabilities if  $C_{IMC}$  is chosen to be used on the true plant. On the contrary, our proposed algorithm ensures

$$\max_{1 \leq k \leq N} \bar{\sigma} \left[ \frac{P_t C_i}{1 + P_t C_i} - \frac{P_t C_i}{1 + P_t C_i} (j\omega_k) \right] \approx 0.1$$

for the maximum achievable bandwidth,  $11.2 \text{ rad/s}$ ;

- the transfer function  $\frac{G_1}{1 + G_1 C_{IMC}}$  has a maximum gain of approximately  $10 \text{ dB}$  since in the algorithm of (Lee *et al.*, 1995) there was nothing limiting this gain but our method gives  $\bar{\sigma} \left[ \frac{P_t}{1 + P_t C_i} \right] \simeq 6.8 \text{ dB}$ ;
- both  $\frac{C_{IMC}}{1 + G_1 C_{IMC}}$  of (Lee *et al.*, 1995) and our  $\frac{C_i}{1 + P_t C_i}$  had a gain less than  $30 \text{ dB}$  and hence with respect to this criterion, both designs were satisfactory;
- for the design of (Lee *et al.*, 1995),

$$\max_{1 \leq k \leq N} \bar{\sigma} [H(P_t, C_{IMC}) - H(G_1, C_{IMC}) (j\omega_k)] = 2.1530$$

whereas in our algorithm

$$\max_{1 \leq k \leq N} \bar{\sigma} [H(P_t, C_{final}) - H(P_3, C_{final}) (j\omega_k)] = 0.257.$$

This shows that the final controller of (Lee *et al.*, 1995) performs differently in a closed-loop sense on  $P_t$  than on  $G_1$ . The difference in closed-loop transfer functions of our final controller on  $P_t$  and on  $P_3$  is well contained.

These points assert the superiority of our algorithm compared to the one of (Lee *et al.*, 1995) in terms of keeping  $\frac{P_t C_i}{1 + P_t C_i}$  close to  $\frac{P_t C_i}{1 + P_t C_i}$ , limiting the size of  $\frac{P_t}{1 + P_t C_i}$  and  $\frac{C_i}{1 + P_t C_i}$ , and ensuring that the flags  $\gamma_i$  and  $\max_{1 \leq k \leq N} \bar{\sigma} [H(P_t, C_i) - H(P_t, C_i) (j\omega_k)]$  are easy to handle.

## 5. CONCLUSION

This paper has introduced a modified algorithm for the windsurfer approach to iterative identification and control by incorporating a new controller design method. This  $\mathcal{H}_\infty$  controller design method can be applied when the plant is stable or unstable, or has  $j\omega$ -axis zeros. Additionally, it is well capable of dealing with plants with lightly-damped poles and zeros and situations where the bandwidth of  $F_i$  is orders of magnitude greater than that of  $P_i$ . This algorithm also provides us with one number,  $\gamma_i$ , that easily flags whether the desired performance specifications have been achieved. Furthermore, an easy computation  $\max_{1 \leq k \leq N} \bar{\sigma} [H(P_t, C_i) - H(P_t, C_i) (j\omega_k)]$  indicates whether re-identification is necessary or not. An extension of this research is currently underway to make the 10% bandwidth expansion rule discussed in Step 6 of the proposed algorithm more automatic and less empirical.

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