

Modelling of near isothermal liquid piston gas compressor employing porous media for compressed air energy storage systems

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Abstract

The liquid piston gas compressor (LPGC) is a method of compressing gases with improved efficiency. Key to the success of this device is its operation in as close to an isothermal state as possible. This paper presents high-fidelity, three-dimensional, unsteady Reynolds-averaged Navier-Stokes (uRANS) simulations to better understand the heat transfer and fluid physics involved in the liquid-piston-driven compression process. Furthermore, the uRANS is coupled with conjugate heat transfer to study using porous media inserts to manage the temperature increase. We simulate the entire cylinder/porous media arrangement using the volume of fluid (VOF) method to analyse the turbulent, multiphase physics and the fluid-structure interaction, providing a greater understanding of this process. It also investigates how porous media inserts perform against the no-insert (baseline) cases in producing a near-isothermal process. The porous mediums used are parallel plates, interrupted plates, and metal foam, all produced from aluminium. Results show that temperature rises within the cylinder can be reduced by as much as 120 K, depending on the choice of porous insert. This temperature reduction translates to an increase of up to 13% in compression efficiency.

Keywords: Computational Fluid Dynamics, Conjugate Heat Transfer, Liquid Piston Gas Compressor, Porous Media, Volume of Fluid

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Nomenclature

Symbols

A	Area	m^2
C	Constant	—
c	Specific heat	$\text{J}/\text{kg} \cdot \text{K}$
Co	Courant number	—
g	Gravity	m/s^2
H	Cylinder height	m
k	Turbulent kinetic energy	m^2/s^2
L	Porous media length	m
p	Pressure	bar
r	Radius	m
Re	Reynolds number	—
T	Temperature	K
t	Time	s
u	Fluid velocity	m/s
V	Volume	m^3
W	Work	J

Greek Symbols

α	Volume phase fraction	—
$\bar{\tau}$	Stress tensor	N/m^2

Δ	Difference	—
ϵ	Dissipation rate for kinetic energy	m^2/s^3
η	Compression efficiency	%
κ	Thermal conductivity	$\text{W}/\text{m} \cdot \text{K}$
μ	Dynamic viscosity	kg/ms
∇	Nabla	—
ρ	Density	kg/m^3
$\sigma_k / \sigma_\epsilon$	Turbulent Prandtl number for k & ϵ , and Energy equation	—

Subscripts

0	Initial
a	Air
ave	Average
c	Compressed
f	Fluid
i	Inlet
in	Input
p	Pressure
s	Solid
s	Stored
t	Turbulent
w	Water

Acronyms

ACAES	Adiabatic compressed air energy storage
CAES	Compressed air energy storage
CFD	Computational fluid dynamics
CO ₂	Carbon dioxide
DCAES	Diabatic compressed air energy storage
GW	Gigawatt
HTC	Heat transfer coefficient
LES	Large eddy simulation
LPGC	Liquid piston gas compressor
PISO	Pressure Implicit with Split Operators
PPI	Pores per inch
SA/V	Surface-area-to-volume
uRANS	Unsteady Reynolds-averaged Navier-Stokes
VOF	Volume of fluid

1. Introduction

The intermittent nature of renewable energy sources presents a significant challenge to electrical grid stability. Energy storage is a critical solution to this challenge, providing the flexibility to store surplus electricity when it is in demand. The majority of recent installations have been deployed as fast-response electricity storage, such as batteries. Currently, the UK operates with 1.6 GW of lithium battery storage, with another 1.5 GW under construction and a further 20 GW in development and planning stages [1]. Albeit mature, this type of technology is classed as a short-term storage method. Additional innovative, longer-term energy storage forms are required to ensure power supply security in all weather conditions (e.g., intermittency of renewable energy generation). Furthermore, this longer-term technology is essential to meet the planned ‘Net Zero’ emissions target of 2050 [2].

One approach that evades the issues associated with ‘shorter-term’ storage strategies is compressed air energy storage (CAES) systems [3]. CAES systems create electricity from the discharge of high-pressure air. Firstly, excess electricity from some external source is used to compress the working gas, which is then sent to the storage facility [4]. Before storage, thermal energy generated by the compression is lost, as the storage of ‘hot air’ is volumetrically inefficient. When power is required, the compressed air is discharged via an expander, which passes through an air turbine to generate the power [5]. The expanding air needs to be reheated by burning fossil fuels. This method has disadvantages due to the CO₂ emissions from burning natural gases in the high-pressure air before they flow into the turbine [4]. The existing plants that use this technology [4, 5] are classed as diabatic compressed air energy storage (DCAES) systems. Another CAES system has been conceived to reuse the heat created during the compression process. This concept is called adiabatic compressed air energy storage (ACAES).

ACAES systems are distinctive in using thermal energy storage (TES) units to capture the heat induced by air compression [6]. The stored thermal energy is transferred back to the expanding air when discharging. This propagation neutralises the adiabatic temperature drop and undertaking this cyclic heat transfer process mitigates the requirement for an auxiliary heat source. Advantages of this type of system are its long-duration electricity storage (days-weeks) combined with a medium-duration discharge (over 4 hours) while providing a high round-trip efficiency (RTE) [7].

One way to improve the efficiency of this ACAES system is to change

38 the compression and expansion method. The liquid piston gas compres-
39 sor/expander (LPGC/E) is a combined compression and expansion unit (Fig-
40 ure 1) that uses a liquid to compress a working gas [8]. The first (compres-
41 sion) stage of the process is completed in the compression chamber before
42 the high-pressure gas is dispatched to a repository. During the second (ex-
43 pansion) process, the recovered working fluid propels the liquid piston back
44 down the chamber. The evacuated fluid passes through a pump and gen-
45 erator to produce electricity. The advantages of this method are that the
46 liquid piston in this combined unit reduces gas leakages and is mechanically
47 simple. The compressor/expander can be fitted with a porous medium inside
48 the chamber to produce a near-isothermal compression/expansion cycle.

49 The key benefit of employing a liquid piston for compressing air is that
50 the porous media in question can be used in the cylinder and reduce the
51 peak air temperature, improving efficiency [9]. The porous media is usually
52 a material with a high thermal conductivity. Porous media with complex
53 geometries usually have a high surface-area-to-volume (SA/V) ratio and sig-
54 nificantly greater heat transfer capabilities than simpler geometries. The
55 additional increase in heat transfer is due to the complex shapes having
56 shorter boundary layers [10]. These shorter boundaries break up the thermal
57 boundary layer (as opposed to the long plates). The complex, varied geome-
58 tries change flow angles, promoting mixing and increasing heat transfer [11].
59 The downfall with these geometries is that they usually come with higher
60 pressure drops [12]. Open-cell metal foams have higher pressure drops than
61 most porous media but are excellent at promoting mixing [13, 14].

62 Metal porous media inserts in compression and expansion is a proven
63 method for increasing LPGC efficiency [15, 16]. At low pressures (10-12
64 bar), interrupted plates and open-cell aluminium foam have been seen to
65 reduce the peak temperature by 90 to 120 K [17]. The baseline case was
66 tested with an empty cylinder, and the efficiency was improved by filling the
67 entire cylinder with porous media. Yan et al., [17] also achieved a reduction
68 of between 57-75% of the baseline (no porous media) temperature, which
69 leads to an 18% increase in compression efficiency.

70 Wieberdink et al., [15] assessed interrupted plates at pressures of up to
71 210 bar. At these higher pressures, the compression process reaches levels of
72 93% efficiency. Also, filling the top 13% of the LPGC with porous media at
73 the higher pressure was as efficient in the lower-pressure case. The significant
74 reason is the porous media engagement with the air and its duration. During
75 compression, the air interacts with the porous media for the entire time



Figure 1: Illustration of liquid piston gas compressor/expander.

76 during both compression and expansion. Therefore, at its highest and lowest
 77 temperature points.

78 Patil et al., [16] investigate the use of different material porous media in
 79 the form of wire mesh. Aluminium and copper wire mesh spirals are fitted
 80 inside a compression chamber and tested at various flow rates. Via this
 81 method, the peak air temperature was reduced by up to 33 K, providing an
 82 efficiency improvement of up to 90%. Overall, this was an 8% improvement
 83 from the baseline cases. Khaljani et al., [9] used five aluminium parallel plates
 84 of various lengths to reduce the peak temperature. By partially filling the
 85 cylinder with five 0.5 m plates, the peak air temperature could be reduced by
 86 up to 80 K from the baseline case (450 K) during a 3.5-second compression
 87 stroke. Furthermore, this temperature reduction improves efficiency by 8%
 88 to 87% from the baseline case (79%). All these cases provide initial evidence
 89 that simple metal inserts effectively improve compression efficiency within
 90 an LPGC arrangement.

91 Multiple other approaches have been attempted to lessen the compressed
 92 air temperature and improve efficiency. Other methods such as micron-sized
 93 spray [18], aqueous foam [19], a high-speed fan [20] and cooling tubes [21]
 94 have been used in an attempt to counteract the temperature rise. Each

95 method has various successes and drawbacks in reducing final air temperature
96 while improving the efficiency of the component.

97 Using a less complicated method, such as aluminium parallel plates, can
98 reduce air temperature and improve efficiency by 16% and 8%, respectively
99 [9]. Using more intricate geometrical features such as foam can reduce the
100 air temperature by 80%, with an associated efficiency increase of 11% [22].
101 Applying more complicated methods, like aqueous foam [19] (efficiency \approx
102 92%), high-speed fans [20] (\approx 95%) and cooling tubes [21] (\approx 99%), all claim
103 to be effective in improving the overall efficiency of the component. However,
104 they come with significant challenges when attempting to implement them
105 at an operational level.

106 Given the relative parity of the differing approaches in terms of their per-
107 formance, there is a case for using the simplest method to implement (i.e.
108 a solid porous insert). Porous inserts reduce the system's complexity while
109 maintaining a high-efficiency rate needed for a near-isothermal compression
110 process. As a more direct method, the inserts reduce the need for mainte-
111 nance due to the lack of moving parts compared with complex methods, such
112 as aqueous foam and sprays. Although complex methods can be applied at
113 a low-pressure range [23] with relative ease, they could become more chal-
114 lenging at higher pressure ranges/ratios and with larger volumes of air to
115 cool. Also, controlling the spray systems could be challenging when scaling
116 up the chamber size, as the studies performed are in small-scale chambers
117 [18, 24]. Furthermore, a key aspect of using the liquid piston is to reduce the
118 number of moving parts and simplify the system's complexity. Using these
119 convoluted methods will only add to this problem. However, these complex
120 cooling methods can enhance the compression efficiency [19]. Nevertheless,
121 applying said methods will create more points of failure within a system than
122 the porous inserts.

123 The existing research on the LPGC with porous inserts has been per-
124 formed using simplified geometry of porous media (parallel plates) [9] or
125 simplified methods of modelling complex porous media cases (metal foam)
126 [25]. Methods such as representative elementary volume (REV) or the porous
127 jump approach, which ignores the pore's intricate flow features, have been
128 used to investigate such cases. This paper will investigate a pore-scale ap-
129 proach to the complex flow and thermal features of an LPGC employing
130 porous inserts. It aims to shed light on the turbulent compressible and
131 multiphase flow in the LPGC using porous media geometries of various com-
132 plexities.

Table 1: Thermophysical properties of LPGC associated parts at $t = 0$.

Part	Material	ρ (kg/m ³)	κ (W/m·K)	c_p (J/kg·K)	μ (kg/m·s)
<i>Working gas</i>	Air	1.205	0.03	1006	1.8×10^{-5}
<i>Liquid Piston</i>	Water	998	0.6	4187	8.9×10^{-4}
<i>Cylinder</i>	Stainless steel	8000	16	500	-
<i>Porous Media</i>	Aluminium	2719	237	871	-

133 2. Methodology

134 In this paper, computational fluid dynamic (CFD) simulations are con-
 135 ducted to establish how each porous media insert performs against the base-
 136 line case (no porous media) in the compression process. The liquid piston is
 137 water, and the working gas to be compressed is air. The porous media cases
 138 (Figure 2) examined are parallel plates, interrupted plates (linear and tilted)
 139 and a ‘foam-like’ lattice structure, all made from aluminium. The plates are
 140 spaced 5 mm apart in the transverse direction, and the interrupted plates
 141 have 5 mm gaps in the flow direction. The foam has pores of 5 mm in diame-
 142 ter, producing 20 pores per inch (PPI). The compression chamber is stainless
 143 steel with dimensions of 0.079 m diameter and 1 m length and a volume of
 144 4.9l. The porous media inserts have an overall diameter of 0.077 m (giving 1
 145 mm clearance between them and the chamber walls), and two section lengths
 146 are considered: 0.2 m and 0.4 m. The 0.2 m length insert is exposed solely
 147 to the airflow displaced by the liquid piston. The 0.4 m length insert is in
 148 contact with both air and water. Initially, the cylinder is filled with 10%
 149 (vol) of water and the remainder with air. Thermophysical properties for
 150 initial conditions ($t = 0$) can be found in Table 1.

151 2.1. Governing equations

152 The simulations investigate the complex features within the compressible,
 153 multiphase turbulent flow and use conjugate heat transfer to determine the
 154 heat transport in both fluid and solid phases. The turbulent flow is modelled
 155 using unsteady Reynolds-averaged Navier-Stokes (uRANS) equations. For

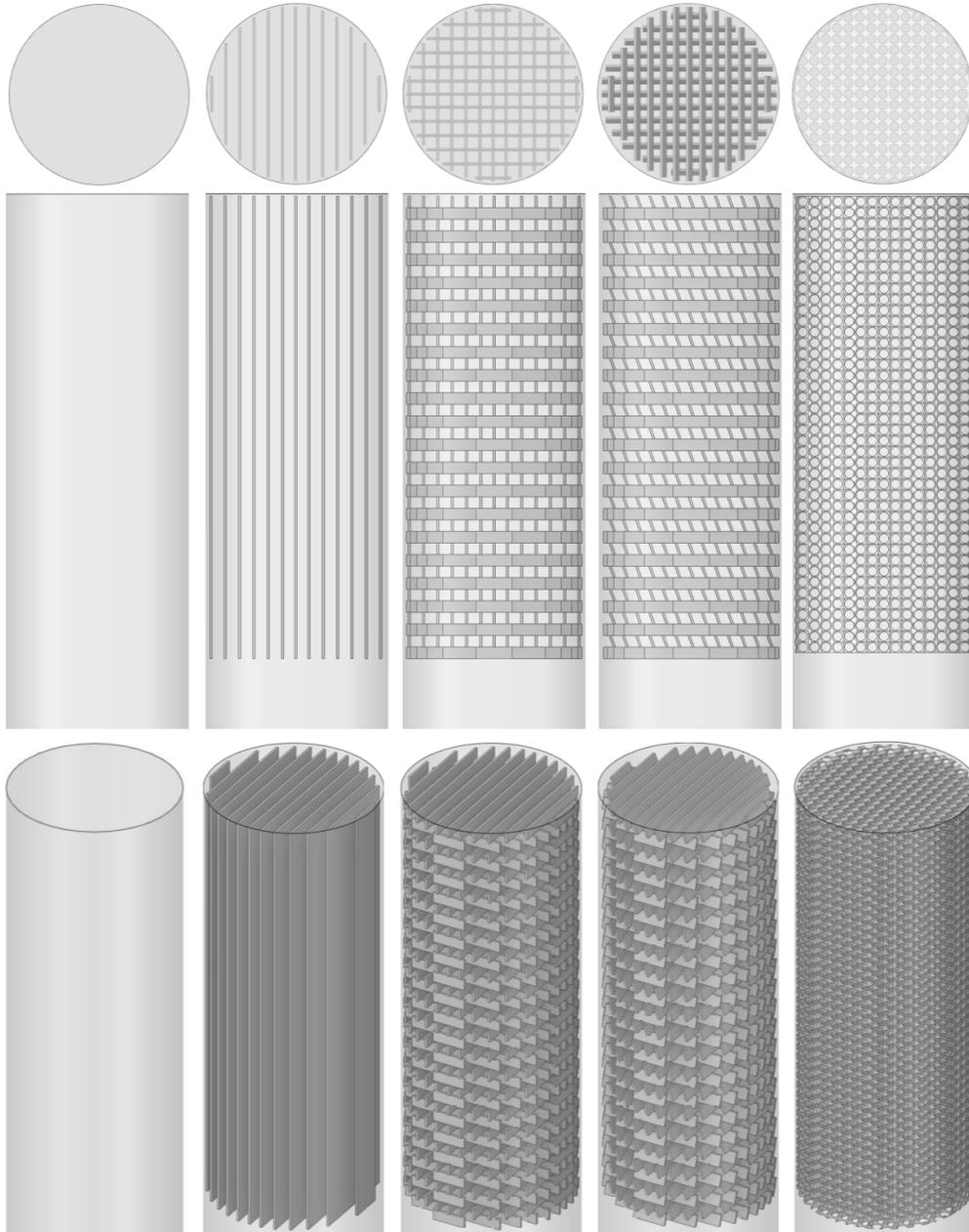


Figure 2: LPGC cases used for CFD simulations with plan, front and isometric views: (L-R) baseline (empty), parallel plates, interrupted plates - linear, interrupted plates - tilted, foam structure.

156 the multiphase problem, the volume of fluid (VOF) method can define the
 157 position of two immiscible fluids by tracking the volume fraction of both
 158 fluids and solving one set of equations. The Eulerian VOF method is applied
 159 in Ansys FLUENT [26] to track the interface between the water and air at
 160 each time step. By starting with the compressible continuity equation:

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho u) = 0, \quad (1)$$

161 we can modify this equation by applying functions of phase volume fraction,
 162 α , to numerically correspond to the transport of both fluid phases. Where
 163 air and water are shown as a and w , respectively. Therefore, the continuity
 164 equations for air is given by:

$$\frac{\partial \alpha_a \rho_a}{\partial t} + \nabla \cdot (\alpha_a \rho_a u) = 0. \quad (2)$$

165 Considering that the density is constant for water (incompressible fluid),
 166 therefore this can be translated into:

$$\frac{\partial \alpha_w}{\partial t} + \nabla \cdot (\alpha_w u) = 0. \quad (3)$$

167 Using the phase-averaged properties, the momentum equation is solved for
 168 both phases via:

$$\frac{\partial \rho u}{\partial t} + \nabla \cdot (\rho u u) = -\nabla p + \nabla \cdot \bar{\bar{\tau}} + \rho g, \quad (4)$$

169 where,

$$\rho = \alpha_a \rho_a + \alpha_w \rho_w, \quad (5)$$

170 and the stress tensor ($\bar{\bar{\tau}}$), which is based on the viscosity of the air-water
 171 mixture, is characterised as

$$\bar{\bar{\tau}} = 2\mu S, \quad (6)$$

172 and

$$\mu = \alpha_a \mu_a + \alpha_w \mu_w. \quad (7)$$

173 Also, using the mean-phase average properties, the energy equation for the
 174 fluid mixture is modelled as

$$\frac{\partial (\rho c_p T)}{\partial t} + \nabla \cdot (u(\rho c_p T + p)) = \nabla \cdot (\kappa_f \nabla T) \quad (8)$$

175 with,

$$\kappa_f = \alpha_a \kappa_a + \alpha_w \kappa_w, \quad (9)$$

176

$$\rho c_p = \alpha_a \rho_a c_{p,a} + \alpha_w \rho_w c_{p,w}. \quad (10)$$

177 For the porous media (solid inserts), the energy equation is combined with
178 the fluid phase as

$$\frac{\partial}{\partial t}(\rho c_s T) = \nabla \cdot (\kappa_s \nabla T). \quad (11)$$

179 The standard $k - \epsilon$ solver is employed for the turbulent flow within the
180 cylinder. For the turbulent kinetic energy (k) and its rate of dissipation (ϵ),
181 the following equations are employed:

$$\frac{\partial(\rho k)}{\partial t} + \nabla \cdot (\rho k u) = \nabla \cdot \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \nabla k \right] + P_k - \rho \epsilon, \quad (12)$$

182

$$\frac{\partial(\rho \epsilon)}{\partial t} + \nabla \cdot (\rho \epsilon u) = \nabla \cdot \left[\left(\mu + \frac{\mu_t}{\sigma_\epsilon} \right) \nabla \epsilon \right] + \frac{C_{\epsilon_1} \epsilon}{k} P_k - C_{\epsilon_2} \rho \frac{\epsilon^2}{k}. \quad (13)$$

183 The shear production of turbulence (P_k) is defined as,

$$P_k = \mu_t S^2, \quad (14)$$

184 with S as the modulus of mean rate-of-strain tensor,

$$S \equiv \sqrt{2 S_{ij} S_{ij}}. \quad (15)$$

185 Furthermore, the turbulent viscosity (μ_t) is determined by the following

$$\mu_t = \rho C_\mu \frac{k^2}{\epsilon}. \quad (16)$$

186 The Prandtl numbers for k and ϵ are given as σ_k and σ_ϵ , respectively, and
187 the turbulent model constants are C_{ϵ_1} and C_{ϵ_2} . These constant parameters
are [26]:

$$\sigma_k = 1.00 \quad \sigma_\epsilon = 1.30 \quad C_\mu = 0.09 \quad C_{\epsilon_1} = 1.44 \quad C_{\epsilon_2} = 1.92$$

188

189 *2.2. Compression efficiency*

190 Improving the heat transfer so a near-isothermal process is achieved dur-
 191 ing the compression will make the process more efficient [15, 16]. Two vari-
 192 ables govern how close the process is to being isothermal: these are the input
 193 work (W_{in}) and the energy stored (E_s). The input work is given as the work
 194 to compress the initial volume of air at ambient conditions (P_0, V_0) to its
 195 final, compressed pressure and volume (P_c, V_c). This is defined by either the
 196 $P - V$ or $T - V$ profile. The input work can be expressed as [17]:

$$W_{in} = (P_c - P_0)V_c - \int_{V_0}^{V_c} (P - P_0) dV, \quad (17)$$

197 E_s is the quantity of work extracted from the compressed air as it is ex-
 198 panded to its initial conditions. If the expansion is slow enough, it occurs as
 199 an isothermal process. If the expansion is fast, an adiabatic process will take
 200 place. In assuming an isothermal expansion, this method quantifies the high-
 201 est potential energy of the reserved cool gas. The storage energy is expressed
 202 as:

$$E_s = P_0 V_0 \ln \left| \frac{P_c}{P_0} \right|, \quad (18)$$

203 where P_c/P_0 is the compression ratio. The compression efficiency is the ratio
 204 of the energy storage to the work input,

$$\eta_c = \frac{E_s}{W_{in}}. \quad (19)$$

205 *2.3. Boundary conditions*

206 An inlet velocity of 0.1785 m/s is used for all simulations leading to a
 207 compression time of up to 3.5 seconds, depending on pressure drop created
 208 by the porous media. The initial pressure and temperature are set to 2.5 bar
 209 and 298 K, respectively. Wall boundary conditions are set for the cylinder
 210 with the adiabatic and no-slip conditions applied. The porous media thermal
 211 settings are applied through conjugate heat transfer within the solver [26].
 212 The air is treated as an ideal gas under compression since pressures are low
 213 (10 bar) temperature differences are relatively small (450 K). The initial
 214 fluid settings are given in Table 1, all values are fixed bar density, which will
 215 vary over the course of the compression. The values are fixed to alleviate
 216 some of the computational cost. For example, dynamic viscosity changes

217 are a function of temperature, by applying Sutherland’s law [27] over the
218 temperature range the change in viscosity would be in the range of $2 \times 10^{-6} <$
219 $\mu < 7 \times 10^{-6}$ kg/m·s. Therefore, this viscosity change and potential effects
220 can be regarded as negligible. Furthermore, the initial surface tension is set
221 at 0.072 mN and the initial contact angle for the interface is 90 degrees.

222 2.4. Simulation parameters

223 First-order implicit methods are used for the transient formulation, and
224 a second-order upwind scheme is used for the spatial derivatives. The PISO
225 (Pressure Implicit with Splitting Operators) algorithm is applied for pressure-
226 velocity coupling. The least squares cell-based method is used to calculate
227 the gradients of variables. All simulations use a time step of 1×10^{-4} seconds,
228 giving a Courant number condition of $Co < 1$. The Courant number ($Co =$
229 $\frac{u_i \Delta t}{\Delta x}$) in this case needed to be less than one due to the transient aspect of
230 the simulations.

231 2.5. Computational meshing

232 The domain is divided into sections with different mesh densities, with
233 finer mesh placed around the inserts and coarser meshing towards the bottom
234 of the cylinder. Depending on the complexity of the inserts, the mesh cell
235 count ranges between 6 million to 30 million elements. As the geometry
236 inside the cylinder becomes more intricate, more elements are required to
237 resolve the flow and thermal boundary layers for these geometries. A y_+
238 value of less than five is imposed around the critical areas of interest for all
239 simulations.

240 2.6. Mesh independence study

241 During the initial stages, domain geometry and grid layout are funda-
242 mental concerns in obtaining a successful simulation solution. A mesh in-
243 dependence study is performed for all simulations to ensure the results are
244 independent of the mesh resolution. The baseline case (Figure 3a and b)
245 and one porous media case (Figure 3c and d) will be reviewed in this study.
246 Mesh independence is conducted on all cylinder/heat exchanger combina-
247 tions. However, only the $h = 0.2$ parallel plate case is shown for the porous
248 media cases to avoid repetition. All values are taken throughout the centre-
249 line of the cylinder. The X-axis plots the dimensionless height within the
250 cylinder (Y/H), 0 being the inlet and 1 being the top wall of the cylinder.

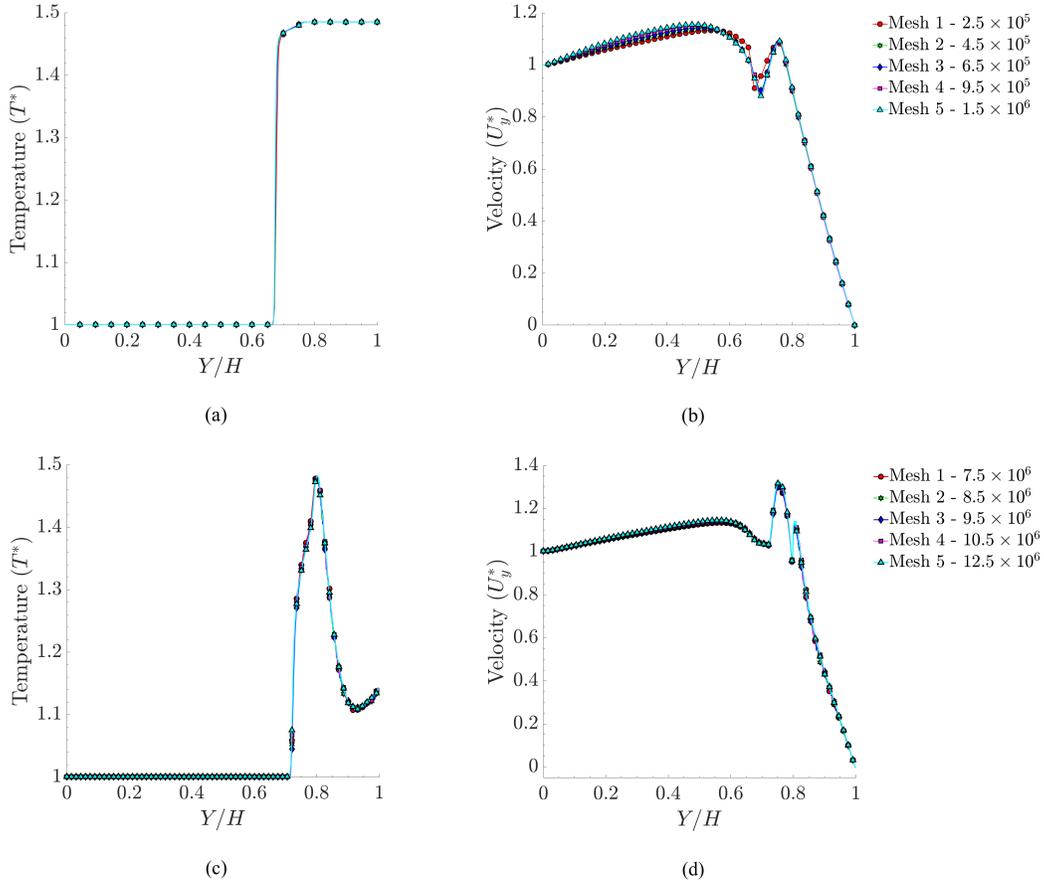


Figure 3: Mesh independence study in which (a) dimensionless temperature (T^*) and (b) velocity (U_y^*) are plotted for the baseline cases. For the porous media case, (c) dimensionless temperature and (d) velocity are plotted. Five meshes are used which increase in refinement from mesh 1 to 5.

Table 2: Dimensionless (normalised) values.

Parameter	Normalised value
Pressure (P)	$P^* = P/P_0$
Velocity ($U_{x,y,z}$)	$U_{\varnothing}^* = U_{\varnothing}/U_0$
Temperature (T)	$T^* = T/T_0$
Turbulent kinetic energy (k)	k/U_i^2
Density (ρ)	$\rho^* = \rho/\rho_0$
Porous media length (L)	$h = L/H$
X-coordinates (X)	X/r
Y-coordinate (Y)	Y/H
Z-coordinate (Z)	Z/r

251 Overall, mesh results do not vary much between all cases. Comparisons
 252 between the fine (mesh 4 and 5) and the medium (mesh 3) resolution meshes
 253 show that the values do not deviate drastically. There are slight variation
 254 between the coarse and fine meshes in both baseline and porous media cases.
 255 However, in both studies the deviations between mesh 3 and mesh 5 can be
 256 classed as negligible so that the changes can be disregarded, especially when
 257 the accuracy difference is minimal. The trade off is a reduction in the element
 258 count of over nine hundred thousand for the baseline; and, 3.2 million for the
 259 parallel plate case. Therefore, no additional refinement is needed as mesh 3
 260 can be demonstrated to be sufficiently accurate for both cases.

261 2.7. Dimensionless variables

262 Dimensionless (normalised) values will be used to compare parameters
 263 with other studies using different liquid piston arrangements. Table 2 dis-
 264 plays the variables, the scaling factor, and the normalised parameter used
 265 throughout.

266 2.8. Validation

267 Experimental data from previous studies [9, 15, 17, 28] are used as a
 268 comparison to validate the current numerical results. Experiments were con-
 269 ducted on empty cylinders comparable to the baseline case in this study.
 270 Figure 4 displays the pressure/temperature-volume charts, which are nor-
 271 malised (Table 2) for ease of comparison. The pressure (P^*) values observed
 272 in all cases follow a similar path in the initial stages, only to vary later due to
 273 the different compression ratios. The present CFD case does not vary much

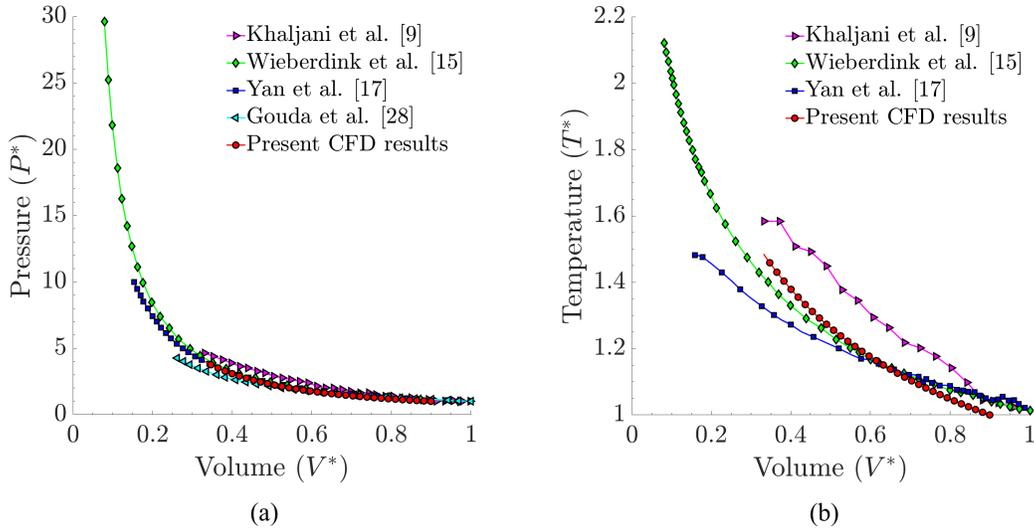


Figure 4: (a) Normalised P - V and (b) T - V comparisons between the current baseline simulation case against various experimental works from Khaljani et al. [9], Wieberdink et al. [15], Yan et al. [17] and Gouda et al. [28].

274 from all the experimental works but is most comparable to the Yan et al.’
 275 [17] study regarding P - V progression.

276 Furthermore, the final T^* values of the numerical results are that of the
 277 Yan study. Also, the Khaljani [9] follows a similar path to the numerical
 278 results. The differences between the Khaljani case can be attributed to a
 279 slight difference in the geometry and experimental variables. The cylinder
 280 height ($h = 0.1$) and compression ratio (four to five) increase in their ex-
 281 perimental [9] study. Changes in said variables can be attributed to the
 282 deviation in temperature developments throughout all studies. During these
 283 baseline compression cases, there is approximately a 6.6% difference between
 284 the numerical and experimental [9] final temperature values. Given this
 285 slight variation, the numerical model accurately predicts the pressure and
 286 temperature increase in this liquid piston arrangement.

287 3. Results

288 This section presents a detailed analysis of the numerical results, with
 289 comparisons made to results from the Neu and Subrenat [29] study, which
 290 analyses the flow fields during a baseline compression process. The results

291 are split into baseline (no porous media) and porous media cases. The base-
292 line cases provide insight into the temperature rise and the flow physics. The
293 porous media cases explore the effects of turbulence on enhancing heat trans-
294 fer. The results for the baseline case are shown over the compression process
295 duration. The porous media cases are given at the end of the compression
296 cycle (10 bar absolute).

297 *3.1. Baseline case (no porous media)*

298 Baseline cases were performed to set benchmarks for the final air tem-
299 perature at the end of the compression cycle and to see how the gas behaves
300 during compression. Figure 5a shows that at the end of the compression
301 process. The average air temperature (T_{ave}^*) of 1.48 (440 K) is reached with
302 a compression time of approximately 3.19 seconds providing the worst-case
303 temperature rise. The mitigation of which is sought via porous media. With
304 this in mind, this empty cylinder case has a compression efficiency of 81.7%.
305 At the onset of the compression process, boundary layer formation is absent
306 at the walls above the water-air interface. Yet, as the compression process
307 advances, the boundary layer undergoes a significant evolution, forming a
308 distinctive funnel-like feature. Despite the use of a fixed inlet velocity (U^*),
309 this asymmetric feature results in a 20% increase in velocity above the water-
310 air interface and along the centreline in the Y-axis (Figure 5b). Which was
311 followed by a progressive reduction in air velocity in the rest of the domain.

312 This effect is further displayed through the radial velocity (U_x^*) above
313 the water-air interface, where these velocities continue to increase as the
314 compression progresses (Figure 5c). This recirculation is initially caused by
315 forced convection as the hot air is compressed, and the cooler cylinder wall
316 evolves into a complex interplay of adverse velocities. Compared to the ex-
317 perimental data, the results align during the initial phases. During the early
318 stages of Neu and Subrenat's [29] work, the flow structure is close to that
319 of our numerical work. With zero velocity at the wall and the gradient in-
320 creasing to the centreline. Furthermore, the opposing radial velocities across
321 the cylinder are similar with ± 0.2 m/s values, which can be seen in both
322 numerical and experimental work.

323 The key difference between this numerical study and the experiment work
324 [29] is capturing how the flow transitions from laminar into a more turbu-
325 lent regime. Our results do not exhibit these turbulent features due to the
326 model's limitations. The standard k-epsilon model calculates the mean flow
327 characteristics via a time-averaging method, which can provide lower fidelity

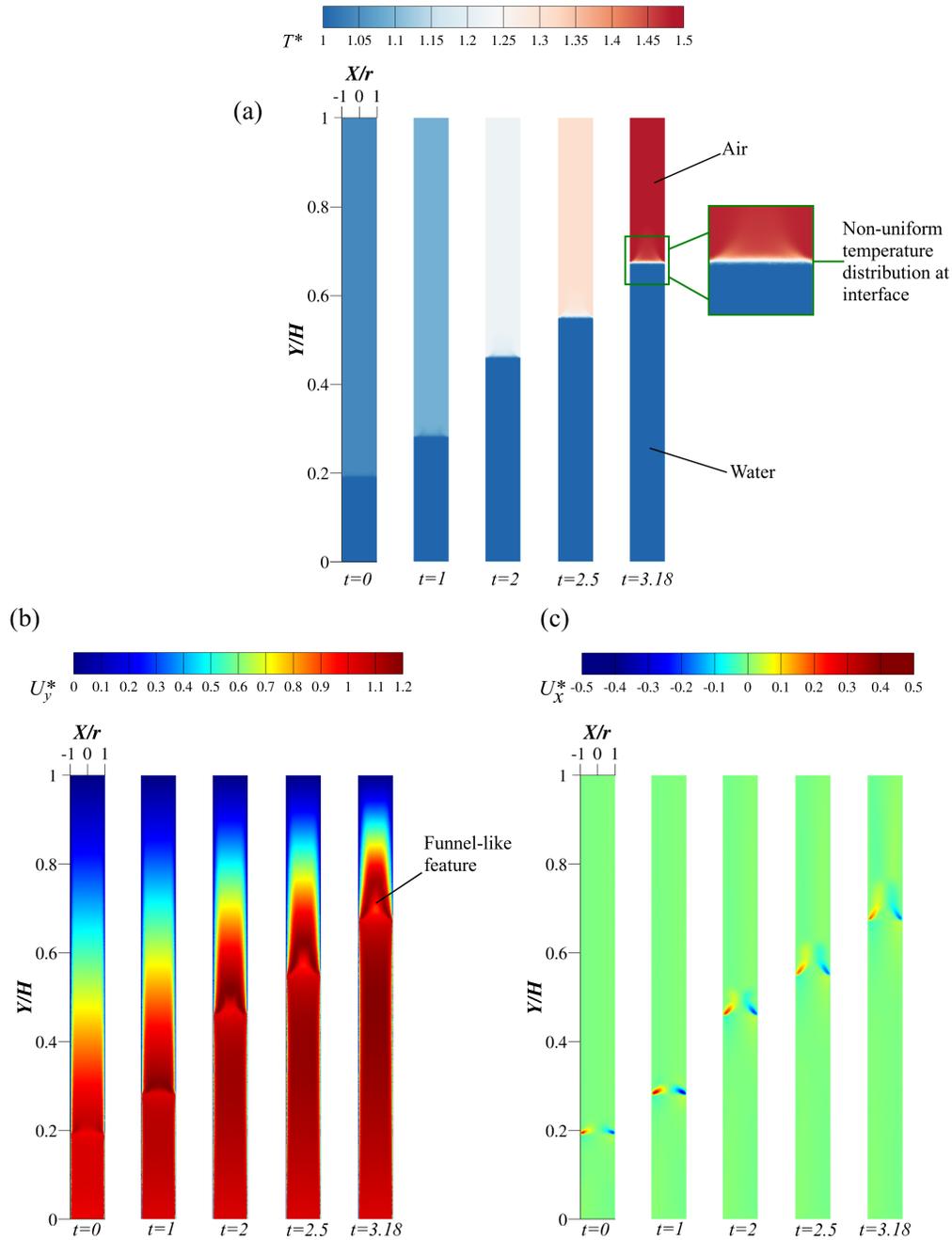


Figure 5: Contours from (a) dimensionless temperature (T^*), (b) dimensionless axial velocity (U_y^*) and (c) dimensionless radial velocity (U_x^*) at different time-steps throughout the compression process.

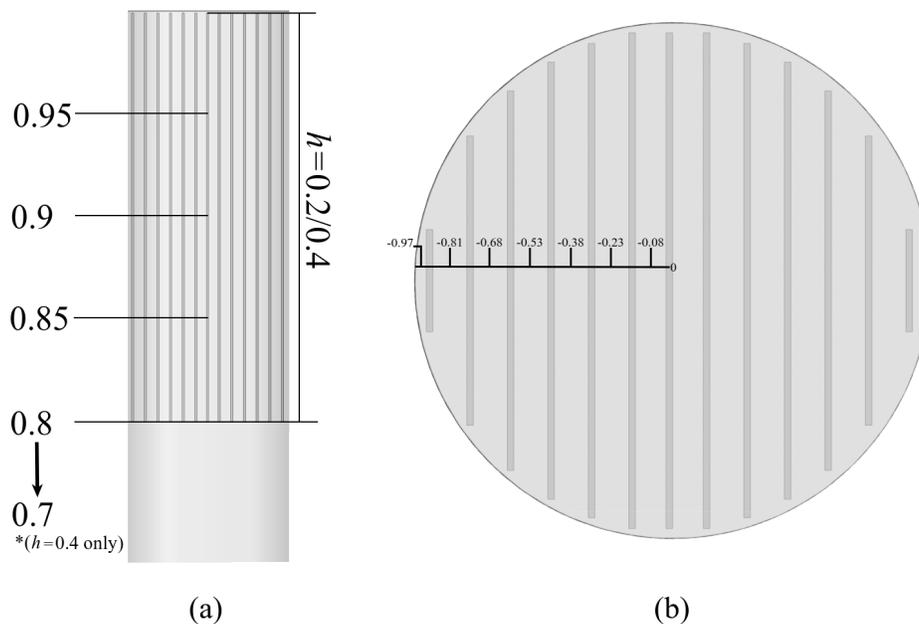


Figure 6: Data extraction locations for (a) Y/H and (b) X/r locations in porous media cases.

328 results when dealing with complicated flow structures. However, higher fi-
 329 delity techniques such as large eddy simulation (LES) modelling have been
 330 used [30, 31] in baseline cases and simple conjugate heat transfer models
 331 (cylinder walls only). LES can capture the intricate flow features with a
 332 higher degree of accuracy by resolving the larger-scale eddies. Also, it pro-
 333 vides more accurate results by accounting for the instantaneous (fluctuating)
 334 flow variables compared to the time-averaged method. However, uRANS was
 335 chosen for this study due to its ability to handle the complexity of adding
 336 porous media without significantly increasing computational costs.

337 3.2. Porous media cases

338 Porous media inserts are essential in producing a near-isothermal process
 339 by increasing the heat transfer between itself and the working gas [15, 17].
 340 Accomplishing this can reduce the final air temperature by increasing the
 341 available heat transfer surface area, improving compression efficiency. In this
 342 paper, we change the arrangement of the porous media to significantly expand
 343 the SA/V ratio, thereby enhancing the heat transfer capabilities within the
 344 system (Table 3). For example, changing the porous media configuration

Table 3: Porous media surface area and volumes.

LPGC cases	SA [m²]	V [m³]	SA/V Ratio
Baseline (no porous media)	-	-	-
Parallel plates: $h = 0.2$	0.304	1.48×10^{-4}	2045:1
Interrupted plates - linear: $h = 0.2$	0.352	1.48×10^{-4}	2371:1
Interrupted plates - tilted: $h = 0.2$	0.367	1.48×10^{-4}	2498:1
Metal foam: $h = 0.2$	0.323	1.03×10^{-4}	3130:1
Parallel plates: $h = 0.4$	0.606	2.97×10^{-4}	2040:1

345 from parallel plates to interrupted plates or open-cell metal foam further
 346 increases the surface area, thus improving the heat transfer between the air
 347 and the porous media. However, this improvement comes at the cost of
 348 increased geometric complexity and higher manufacturing costs, particularly
 349 when transitioning from parallel plates to metal foam.

350 Another parameter tested is the porous media length. The effects of
 351 the insert length were analysed for the parallel plate case using lengths of
 352 $h = L/H = 0.2$ and $h = 0.4$. Placing plates with a length of $h = 0.2$
 353 inside the compression chamber allows the porous media to interact with the
 354 air exclusively. This allows for a higher volume of air to be compressed. By
 355 extending the plate length to $h = 0.4$, the elements become partly submerged
 356 in the water, which will reduce the further temperature by cooling the plates.

357 Also, Wieberdink et al., [15] found that filling only the top half of the
 358 cylinder (air section) provides similar efficiency to cases where the porous
 359 media extends the entire cylinder length. Compared to the full-length con-
 360 figuration, partially filling the chamber offers several advantages. Partial
 361 filling allows more air to occupy the chamber, enabling the compression of
 362 a larger volume of air per stroke. A shorter porous media design decreases
 363 material costs while contributing to a lighter overall system. For these rea-
 364 sons, this study opted for inserts that filled only the top air section of the
 365 cylinder.

366 Furthermore, turbulence levels are analysed within the porous media. By
 367 simulating a whole porous media arrangement rather than REV, turbulence
 368 is investigated without using some form of average values to calculate the
 369 Reynolds (Re) number. Using a normalised turbulent kinetic energy (k/U_i^2),
 370 we can visualise how localised flow disturbances caused by the porous media

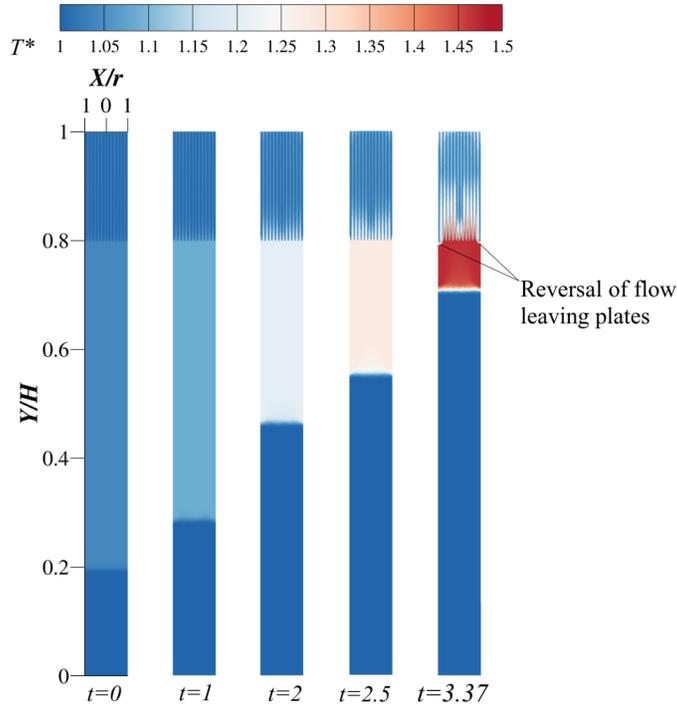


Figure 7: T^* distribution for the parallel plate case at a length of $h = 0.2$ during different time-steps throughout the compression process. Time of 3.37 seconds corresponds to the end of the compression process.

371 affect the heat transfer. For our results, data is taken at Y/H positions, which
 372 span the cylinder's diameter. Furthermore, the X/r data positions are taken
 373 at a radial point and through the length of the porous media inserts (Figure
 374 6).

375 3.2.1. Parallel plate cases at lengths $h = 0.2$ and $h = 0.4$

376 In this case, by placing parallel plates in the cylinder, the final air T_{ave}^*
 377 is restricted to around 1.12 (332.75 K) from the initial value of 1 (298 K).
 378 The compression process takes approximately 3.36 seconds with a pressure
 379 drop ($\Delta P/h$) of approximately 95.8 Pa/m. Added friction from the porous
 380 media inserts creates this differential in pressure. During compression, this
 381 reduction in the air temperature is observed the higher the air travels verti-
 382 cally through the porous media inserts. Figure 7 shows that this temperature
 383 reduction is distributed non-uniformly within the plates. This ranges from
 384 1.33 (396 K) from the base to 1.15 (342 K) at the top of the cylinder, giving

385 the air a 0.18 (54 K) variation across the plates. The average plate T^* rise
386 is just over its initial value of 1.

387 During this, the higher heat transfer coefficient (HTC) values occur at
388 the stagnation region at the plate entrance. This value gradually decreases,
389 moving towards the trailing edge of the porous media. With a slight increase
390 is observed at the top of the plates. This increase in heat transfer is localised
391 in the areas of higher levels of turbulence (Figures 9b and 10a). At the
392 cylinder wall, the boundary layer above the water-air interface grows. Adding
393 the plates creates a further differential as the air enters and travels up these
394 thin channels. A channelling effect occurs when the air enters the porous
395 media and U_y^* increases significantly. Turbulent kinetic energy levels are
396 boosted as U_y^* increases. When the boundary layer approaches $X/r = 0$ on
397 both sides, the velocity sees a 20% reduction from the initial value when
398 entering the plates.

399 However, figures 7 and 8 show that the temperature reduction decreases
400 as the velocities increase. The elongated feature of these plates will create
401 an extended thermal boundary layer (Figure 9) over the length of the plates.
402 Convective heat transfer is restricted due to the thickness of this thermal
403 boundary layer. Also, a negative U_y^* value between the cylinder wall and the
404 neighbouring plate is observed as the reversed flow leaves the porous media.
405 Along with the recirculation, a further narrowing between the final plate and
406 wall and the added convection causes this reversal. To sustain the turbulence
407 seen at the plate entrance, the geometry must be optimised to maintain these
408 levels while reducing the thermal boundary layer.

409 Results for the parallel plates of $h = 0.2$ are similar to Khaljani et al.' [9]
410 experimental study regarding temperature reduction. They use five plates of
411 the same length to achieve a T^* reduction of 0.16 (50 K). Compared to the
412 thirteen in this case, which reduces the air temperature by 0.36 (107.49 K).
413 A 2.4 times plate increase leads to a 2.15 times reduction in air temperature.
414 Increasing the plate quantity displays the interrelationship between expand-
415 ing the surface area and transferring heat in the LPGC. Also, widening the
416 plate widths will affect turbulence production within the plates.

417 Extending the porous media length inside the cylinder is another promis-
418 ing avenue for enhancing overall heat transfer. This approach substantially
419 improves heat transfer by many factors. Firstly, the insert surface area is
420 increased for heat transfer. Doubling the area increases the HTC (section
421 interacting only with air) to 246.83 W/m²K—an improvement of 59% from
422 the $h = 0.2$ case. Secondly, submerging the plates in water plays a significant

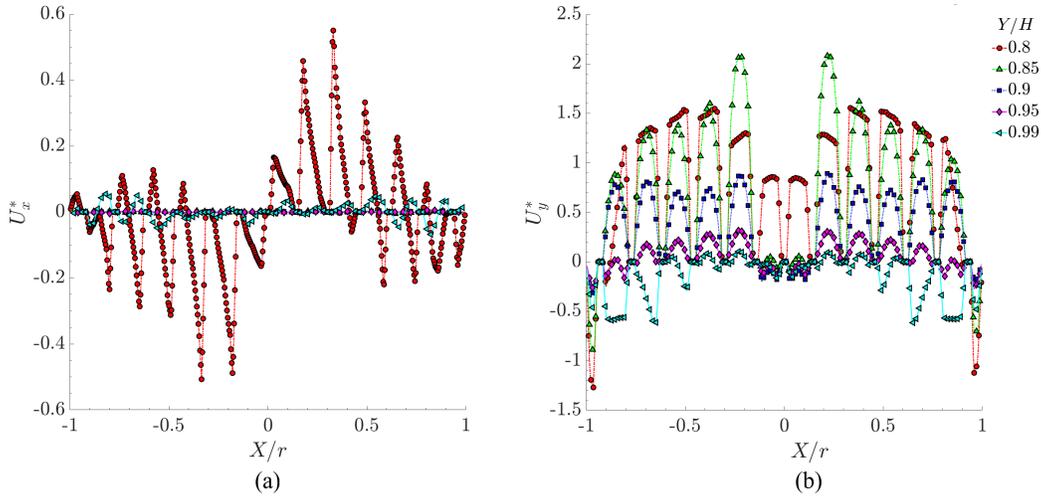


Figure 8: (a) U_x^* and (b) U_y^* velocity components in the in parallel plate case at length of $h = 0.2$ at different Y/H positions.

423 role in improving efficiency. The water reverses the heat flux in the bottom
 424 section of the plates, adding a cooling effect. Nevertheless, the pressure drop
 425 trade-off arises again as this increases to 3354.9 Pa/m. The presence of water
 426 within the extended inserts requires further work to overcome this loss due
 427 to the added viscosity. Even though this pressure drop figure increases from
 428 the shorter inserts, a compression time of 3.42 seconds is achieved.

429 From a turbulence production perspective, the submergence is a nega-
 430 tive. Figure 12 displays how immersion of the insert significantly diminishes
 431 turbulent effects. Even though U_y^* increases by 50% due to the channelling
 432 effects from the narrow gaps, U_x^* and turbulent kinetic energy are greatly re-
 433 duced. Still, the temperature increase is significantly suppressed due to the
 434 emphasis on increased SA/V ratio and negative heat flux. A reduction to the
 435 overall T_{ave}^* temperature is just above 1.07 (320 K). Figure 11 demonstrates a
 436 uniformity in the temperature distribution throughout, with a slight temper-
 437 ature increase of near 1.25 (372 K) in the regions with increased turbulence
 438 (Figures 11b and 12c). Also, the air is contained within the porous media due
 439 to the plates being submerged. This development eliminates all the higher-
 440 temperature air from the $h = 0.2$ cases between the water-air interface and
 441 the plate entrance.

442 Furthermore, the efficiency of the parallel plates with a $h = 0.4$ is the
 443 highest at over 93%. Increasing approximately 6% from the $h = 0.2$ plates

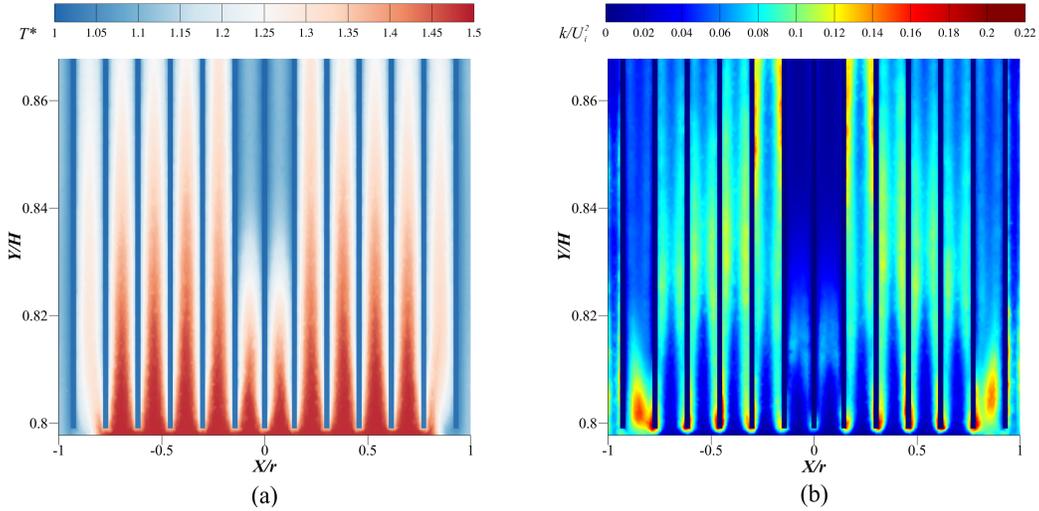


Figure 9: (a) Normalised temperature (T^*) and (b) turbulent kinetic energy (k/U_i^2) distribution in parallel plate entrance in length of $h = 0.2$. Both variables displaying the formation of secondary boundary layer over the plates.

444 and over 12% from the baseline case. The subsequent sections will exam-
 445 ine the effects of the modified porous media geometries on the compression
 446 process.

447 3.2.2. Interrupted plates (linear and tilted) cases at the length $h = 0.2$

448 Refining the geometry by decreasing the lengths and stacking the plates
 449 expands the surface area by 14.76% (parallel plates: $h = 0.2$). The geome-
 450 tries do not vary volumetrically but altering the profile of the porous media
 451 creates better heat transfer attributes from parallel to interrupted plates.
 452 Thus, increasing the SA/V ratio further improves the heat transfer abilities
 453 of the inserts. Altering the geometry breaks up the thermal boundary layer.
 454 Regardless, compression time is slightly increased to 3.39 seconds with an
 455 increased pressure drop of 103.67 Pa/m. Although the linear interrupted
 456 plates still produce a channel flow, this structure also promotes mixing by
 457 splitting up the viscous sub-layer of flow with the shorter plates. Analogous
 458 to the parallel plates, the temperature reduction is non-uniform. Still, this
 459 porous media type exhibits significantly less high-temperature air towards
 460 the base and less temperature variability inside than the parallel plates of
 461 $h = 0.2$. Figure 13 shows that by changing the arrangement of the plates
 462 and using interrupted plates in different arrangements, T_{ave}^* can be further

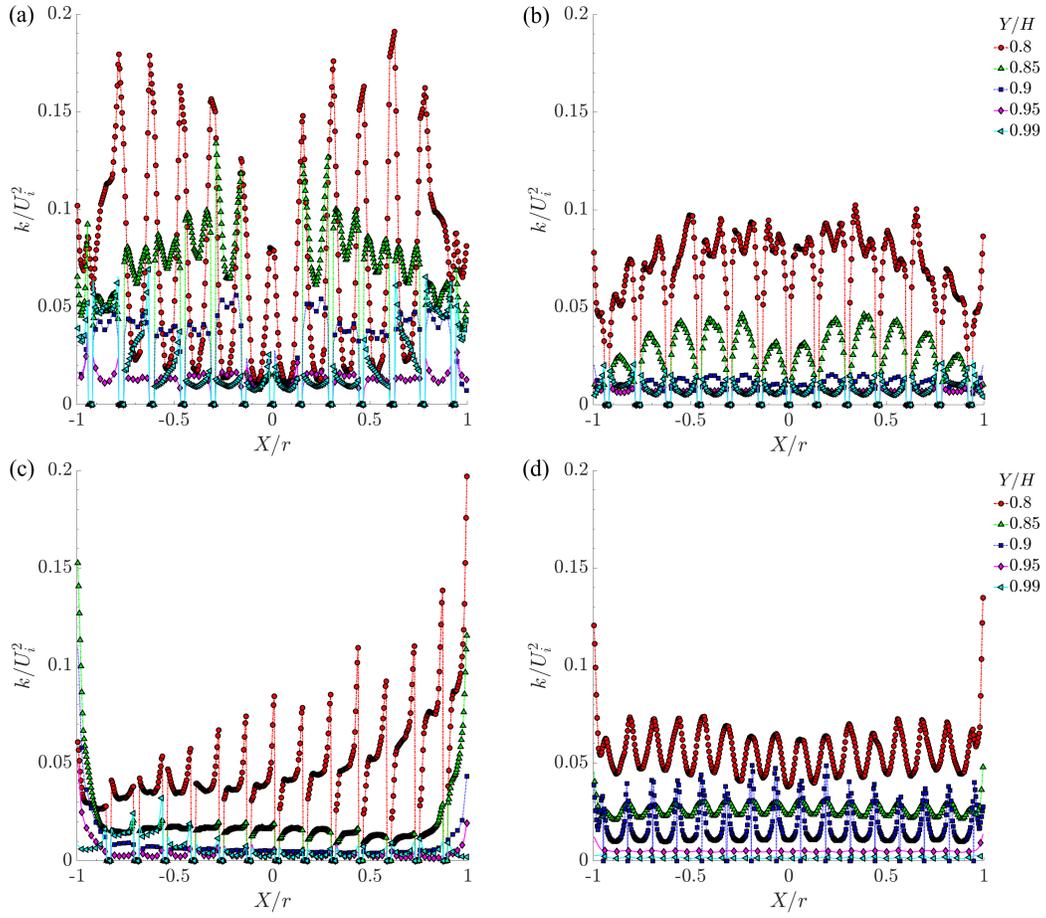


Figure 10: Non-dimensional turbulent kinetic energy (k/U_i^2) values for all porous media types at $h = 0.2$. Figures are taken end of compression cycle (approximately 3.4 seconds). Porous media type: (a) parallel plates, (b) linear interrupted plates, (c) tilted interrupted plates, (d) foam structure.

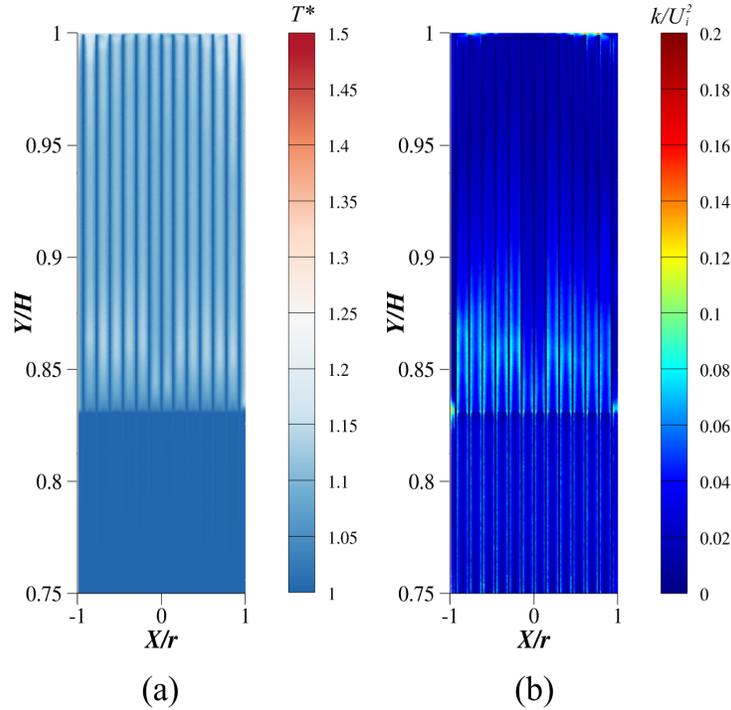


Figure 11: (a) Temperature (T^*) and (b) turbulent kinetic energy (k/U_i^2) distribution in parallel plate case at length of $h = 0.4$. Figures taken end of the compression process (3.42 seconds).

463 reduced to around 1.1 (326.64 K).

464 As the flow passes through the channels, it mixes freely in each subsection.
 465 The production of vortices can be seen in each longitudinal section (Figure
 466 14), where the density difference across the plates is at its highest. From
 467 the entrance of the cylinder, ρ^* increases from around 6.5 to 9.5 between
 468 0.9-0.95 Y/H mark, which is approximately a 46% increase. This originates
 469 from the temperature disparity between fluid and solid. Also, viscous/form
 470 drag comes into effect from the plates, which can also be characterised as
 471 between Darcy and Forchheimer-type flows ($\approx 10 < Re < 150$). Stated flows
 472 border on laminar to unsteady laminar; this is evident in Figure 10b in which
 473 turbulence levels are non-existent between these Y/H values.

474 Furthermore, tilting the plates should increase heat transfer capabilities
 475 theoretically; this is due to the expanded SA/V ratio (5.6% increase). The
 476 tilt angle was selected following the Zhang [32] study. Zhang found that
 477 tilting the angle of the plates between 0 and 40 degrees increases the heat

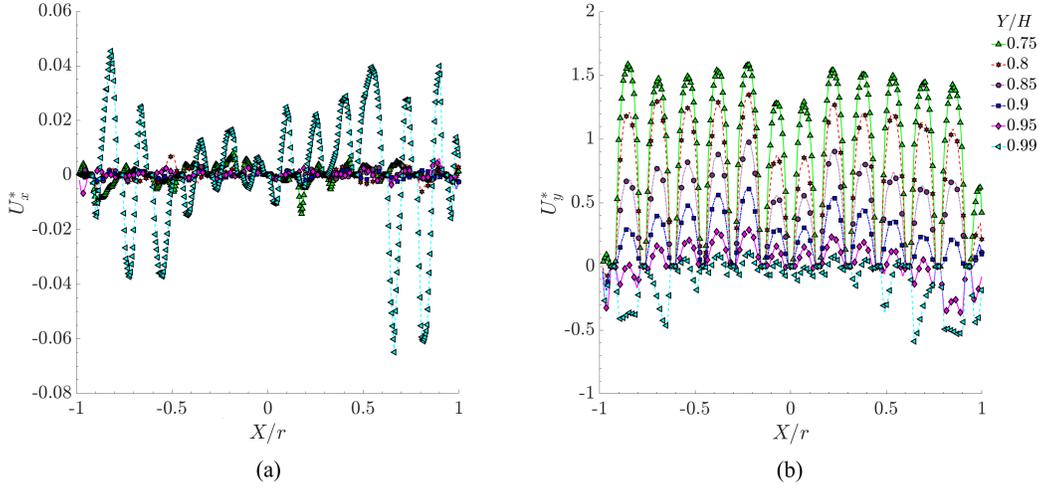


Figure 12: (a) U_x^* and (b) U_y^* velocity components at different Y/H positions of parallel plate case at length of $h = 0.4$. Figures taken at end of compression cycle (3.42 seconds).

478 transfer. The increased impingement on the plates leads to a progressively
 479 intricate pore-scale fluid flow. This geometric change increases turbulence
 480 and, therefore, heat transfer. Therefore, a median angle of 20 degrees in the
 481 X- and Z-axes (Figure 2) was selected for this study. However, Zhang's [32]
 482 study only implemented REV methods for simulating this type of setup, in
 483 which the full effects of this plate arrangement are not observed.

484 The flow is forced into the direction of the tilt, as this is 20° from left to
 485 right, creating a spiralling effect through the porous media. Evident in the U_x^*
 486 and U_y^* velocity profiles in Figure 15. After the initial entry to the tilted plate,
 487 as per the linear plates, the heat transfer becomes more uniform throughout
 488 both porous media. Figure 13 illustrates that even with these added heat
 489 transfer enhancements, the temperature distribution through both sets of
 490 interrupted plates is nearly identical. In both cases, the ΔT between the air
 491 T_{ave}^* and the plate temperature are negligible. Again, with a higher surface
 492 area, frictional forces increase to raise the pressure drop to 109.5 Pa/m.
 493 Leading to a compression time of 3.43 seconds.

494 Figure 10c displays close to a three-to-four times increase in turbulence
 495 kinetic energy in the tilted plates in localised areas that follow the tilt direc-
 496 tion. This increase is also observed in the U_y^* velocity profile, and the HTC
 497 peaks at around 1360.79 W/m²K. Although there are fluctuations in localised
 498 areas, the tilt reduces the overall turbulence levels throughout the insert. If

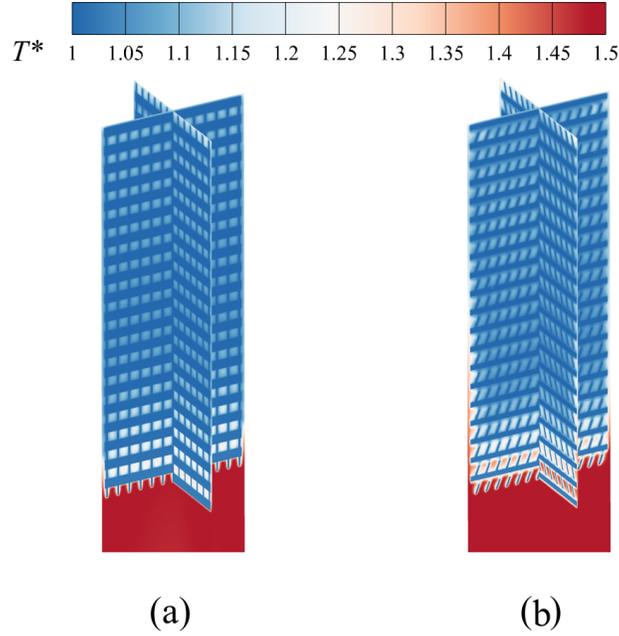


Figure 13: T^* distribution for the (a) interrupted and (b) tilted plate cases at a length of $h = 0.2$.

499 the tilted geometry and varying angles can be optimised to improve overall
 500 turbulence levels, this could be a promising avenue regarding heat transfer
 501 in this application. Furthermore, creating a swirling flow can increase heat
 502 transfer by fifty percent compared to the non-swirling flow with identical in-
 503 let conditions [33]. In this case, the overall HTC of the tilted plates increases
 504 by approximately 11.8% (22 W/m²K) over its linear interrupted counterpart
 505 (173.25 W/m²K). Still, this added heat transfer efficiency only improves by
 506 0.45% between the linear (87.95%) and tilted (88.4%) versions.

507 3.2.3. Metal foam case at a length of $h = 0.2$

508 Lastly, this geometry modification changes the structure to one that is
 509 not linear in its features as per the other inserts. A “foam-like” lattice
 510 structure was created to enhance the SA/V ratio and turbulence further.
 511 The foam holds the highest SA/V ratio of 3130:1. It averages a HTC of
 512 265.21 W/m²K, the highest of all the porous media cases. The HTC increase
 513 is due to higher heat flux and a reduction in ΔT between the insert and air.
 514 It is further acknowledging the importance of this variable in heat transfer.
 515 Turbulent kinetic energy and HTC levels are highest through the initial plate

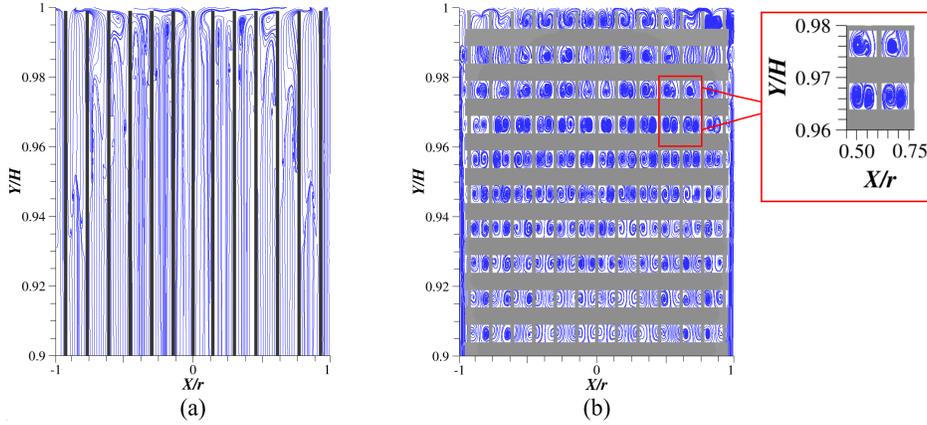


Figure 14: Streamlines within top 10% section of the (a) parallel plate and (b) linear interrupted plate cases at a length of $h = 0.2$.

516 entrance. Turbulence remains steady until around the $Y/H = 0.85$ mark
 517 (Figure 10d). The increase in turbulence and velocity is a direct result of
 518 the increased surface area and the pores across the bottom face of the insert.
 519 Figure 16 shows an U_y^* increase to approximately 1.6 as it passes through the
 520 plate/pore entrance, U_x^* velocity exhibits high values again at this point only.
 521 As the air passes through with this increased velocity through the pores, the
 522 temperature reduction is minimal.

523 Figure 17 displays that as the air enters the insert at a T^* of slightly
 524 less than 1.5, it is drastically cooled and is reduced to slightly over one
 525 towards the top of the cylinder. The overall T_{ave}^* reduces to approximately
 526 1.1 (328.19 K). The higher the porous media the air travels, the more all these
 527 values (T^* , U^*) are reduced significantly. As per the interrupted plates, the
 528 air temperature distribution is more uniform than the parallel plates. An
 529 average temperature increase of 0.101 (30.1 K) from the initial baseline cases
 530 is further evidence for a near-isothermal process using this porous media
 531 type.

532 However, compression time rises to 3.48 seconds with a further pressure
 533 drop of 113.29 Pa/m. This pressure drop might not be significant during
 534 a fast flow rate case, as the compression only rises 0.28 seconds from the
 535 baseline case, therefore not requiring significant extra work. Once coupled
 536 with a longer compression time, the effects of this pressure drop may become
 537 detrimental to the amount of work needed to compress the air. Neverthe-
 538 less, changing the structure to a foam lattice shape improves efficiency up

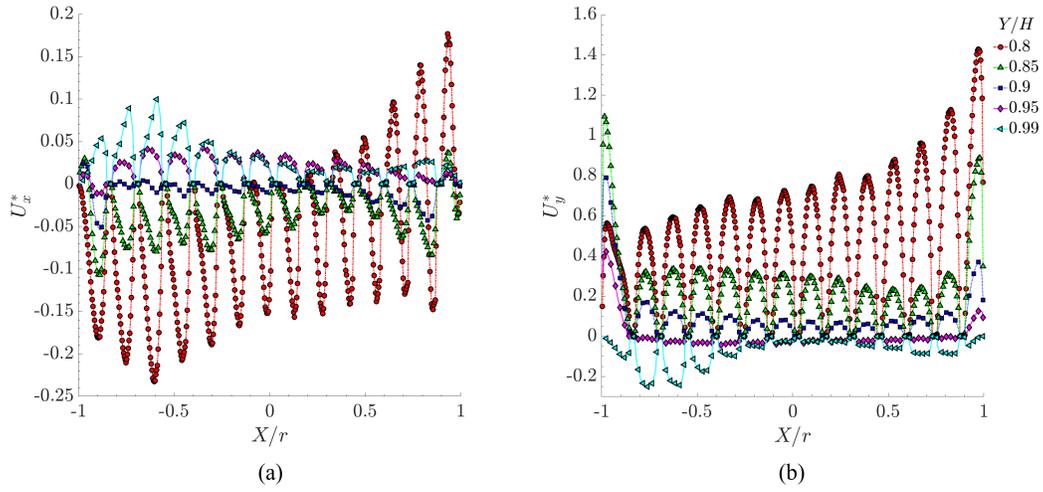


Figure 15: (a) U_x^* and (b) U_y^* velocity profiles for the tilted plate case of $h = 0.2$.

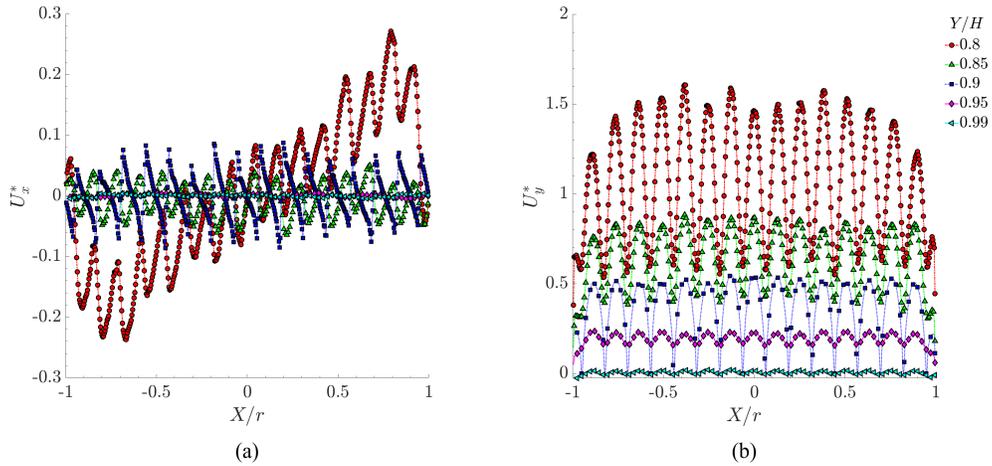


Figure 16: (a) U_x^* and (b) U_y^* velocity components at different Y/H positions in the metal foam case of $h = 0.2$.

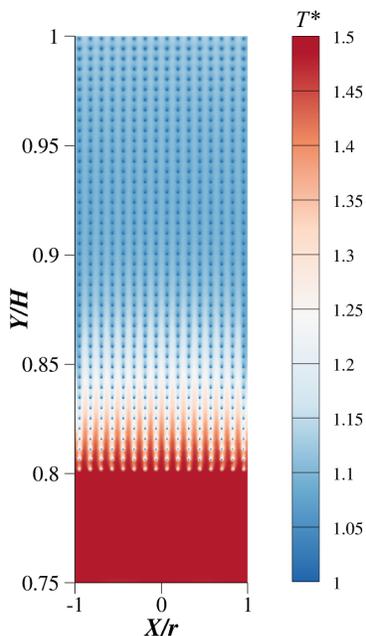


Figure 17: T^* distribution at end of compression process in the metal foam case of $h = 0.2$.

539 to approximately 89%. However, this is only a 1% difference from the lin-
 540 ear interrupted plates. The lattice structure is still relatively structured
 541 compared to metal foams from other studies. Future work is needed to opti-
 542 mize the porous media to increase turbulence through stochastic methods
 543 [34] to improve the HTC. Therefore, it is an additional enhancement to the
 544 compression process. Regarding thermal management, the trade-off between
 545 porosity and pore size is the crucial aspect [13]. Nevertheless, with a value of
 546 approximately 89%, this is a good starting point to improve these structures.

547 4. Conclusion

548 A compressible, multiphase, turbulent heat transfer analysis was per-
 549 formed to investigate the heat transfer and fluid flow within a Liquid Piston
 550 Gas Compressor. The LPGC compresses air at ambient temperature from
 551 2.5 to 10 bar. The primary focus of this study was to increase the heat
 552 transfer within the cylinder by using porous media inserts. Four different
 553 types (at two lengths) of porous inserts were used. Geometric changes to
 554 the insert were made to evaluate various parameters such as the SA/V ratio,
 555 turbulence properties and the effect of water cooling. The increased heat

556 transfer with the porous media for the compression process reduces the final
557 air temperature to as near isothermal as possible. Therefore, this increase in
558 SA/V ratio and turbulence enhancement improves the overall compression
559 efficiency. Further geometry optimisation is needed to take full advantage of
560 turbulence production within the inserts rather than just elevated levels at
561 the entrance.

562 Coupling the extended length with more intricate geometries can increase
563 compression efficiency, but a significant trade-off is required for these more
564 extended inserts. With a performance improvement, there is a reduction in
565 the volume of air to be compressed. On a relatively small-scale component
566 such as this five-litre unit, is the extra 4% performance worth the loss in
567 compressed air. Nevertheless, the pressure drop produced by the extended
568 inserts creates another limitation to overcome.

569 Further preliminary investigations are ongoing concerning how reduced
570 flow rates and compression time can affect the heat transfer within the porous
571 media. The current results take up to 3.5 seconds to complete a compression
572 cycle. Slowing this process gives the porous media more time to interact with
573 the air. Furthermore, this work will investigate how both fluids interact with
574 the plates and if any turbulence is featured within the complex geometries.
575 Extending the complex geometries to this length will further reduce overall
576 air temperature. Further breaking up the thermal boundary layer to allow
577 for more mixing will enhance the heat transfer abilities. Also, following the
578 trend in Table 3, the SA/V ratio will increase, further enhancing efficiency.
579 It is essential to note that the issue with these longer porous media is the
580 increased mesh sizes, which will be more computationally expensive than the
581 $h = 0.2$ cases.

582 Furthermore, future work will be undertaken on how both compression
583 and expansion processes using porous media will affect the round-trip effi-
584 ciency (RTE) if implemented in an ACAES system. Said work would be
585 comparable to the Gouda et al.' [35] study in which the RTE of a baseline
586 case is investigated. Employing a similar approach but applying the method-
587 ology from this study or a higher fidelity model will provide greater insight
588 into the LPGC/E effectiveness if implemented into an ACAES system.

589 **Declaration of competing interest**

590 The authors declare that they have no known competing financial inter-
591 ests or personal relationships that could have appeared to influence the work

592 reported in this paper.

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